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GEGENSCHEIN ORBITAL PARAMETERS AND OPERATIONAL SCHEDULE

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STEPHEN J. PADDACK

June, 1964

Goddard Space Flight Center Special Projects Branch Theoretical Division

ABSTRACT

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There are two basic purposes for this document. One is to present and apply a technique for defining the location of the Eccentric Orbiting Geophysical Observatory (EGO), S-49 in the gegenschein reference frame. The gegenschein reference frame is an orthogonal coordinate system with its origin at the center of the earth and its fundamental plane lying in the ecliptic plane with one of the axes in this plane pointing directly away from the sun.

A second purpose for this document is to predict an operational schedule for the gegenschein experiment. The experiment device cannot tolerate direct or reflected sunlight. The gegenschein experiment package is placed on the darkside of the solar array, therefore, direct sunlight is not a problem. However, the earth, moon or appendages of the satellite itself may come into the field of view of the gegenschein experiment. When this occurs the experiment will have to be turned off. A technique for computing when any of these reflecting bodies come into the field of view of the experiment is developed here and applied.

The position of the satellite in the gegenschein coordinate system is a function of the orbit and position of the sun. The field of view problem is a function of the location of the satellite, earth and moon and solar array position; the latter because the gegenschein experiment package rotates with the solar array.

As a result of the application of these techniques, in computer programs, the position in the gegenschein coordinate system versus time of the EGO is presented and also an operational schedule as a function of solar array angle.

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а Semi-major axis. Radius of the earth. $C_{\mathbb{H}}$ Position matrix whose elements are the components in the equatorial coordinate system. $^{\text{C}}\varepsilon$ Position matrix whose elements are the components in the ecliptic coordinate system. Position matrix whose elements are the components in the gegenschein coordinate system. Eccentricity. е i Inclination. \bar{i} , \bar{j} , \bar{k} Orthogonal unit vectors. \overline{N} Unit normal vector. r Position vector. Т Transformation matrix. īī Unit vector along position vector. Cartesian coordinates or component along i, j, k. х, у, г Earth angle. The angle between the anti-solar direction and a vector pointing to the earth. В Moon angle. The angle between the anti-solar direction and a vector pointing to the moon. Experiment angle, angle between the normal to the gegenschein and a vector to an experiment. Angular dimension of earth. Obliquity of the ecliptic. Α Angular displacement between ecliptic and gegenschein coordinate systems. Solar array angle. φ Right ascension of the ascending node. Ω

Argument of perigee.

Œ

Subscripts

c	Corrected.
E	Equatorial coordinate system.
g	Gegenschein coordinate system.
i	index, $i = 1,2,3n$.
ig	Gegenschein experiment to the ith experiment.
m	Earth to moon.
mv	Vehicle to moon.
s	Earth to sun.
v	Earth to vehicle.
sv	Vehicle to sun.
€	Ecliptic coordinate system.
0	Initial conditions.
1	First transformation.
2	Second transformation.

INTRODUCTION

The gegenschein phenomenon is a faint nebulous light which appears in that part of the sky which is opposite the sun. Hence, it rises when the sun sets and reaches its zenith at midnight. It seems to lie in the plane of the ecliptic and stretches out about 10° on either side of the antisolar direction. The gegenschein is sometimes referred to as counterglow.

An experiment is planned to make measurements of intensity and location of the gegenschein. Dr. C. L. Wolff and Dr. K. L. Hallam of the Goddard Space Flight Center and Prof. Wyatt of the University of Illinois have a device on the Eccentric Orbiting Geophysical Observatory (EGO, S-49) which will carry out these measurements.

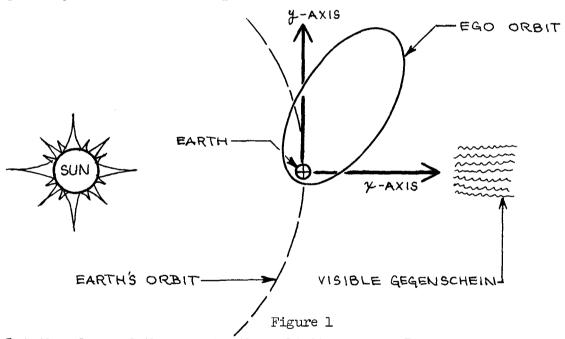
The EGO is the first in the series of Orbiting Geophysical Observatories, the OGO Program. It will have an apogee height in the order of 80,000 nautical miles. The primary objective of the OGO Program is to conduct large numbers of significant, diversified experiments. The EGO will carry twenty experiments, one of which is the gegenschein experiment (Reference 1).

The purpose of this discussion is to show the development of a "gegenschein coordinate system" and a technique for calculating certain angles in space and on the spacecraft.

I. GEGENSCHEIN COORDINATE SYSTEM

Since the gegenschein lies opposite the sun, with respect to the earth, it is convenient to have a coordinate system such that the x-axis points toward the gegenschein. If the position of the satellite is known in this reference frame then measurements of the gegenschein by the satellite will define the position of the gegenschein in space in the gegenschein coordinate system. These measurements can then be transformed into a more conventional reference frame, such as ecliptic or equatorial coordinates.

Consider the sketch below for a simplified representation of the geometry involved with the problem.



Let the plane of the page be the ecliptic plane. The plane formed by the earth's orbit around the sun is the ecliptic plane. Then the plane formed by the x and y axis in the sketch lies in the ecliptic plane. Consequently, the z-axis of the gegenschein coordinate system

is normal to the ecliptic plane. Since the orbit of the EGO does not lie in the ecliptic plane the satellite will have, in general, a z component in the gegenschein coordinate system.

When the satellite is in orbit the position of the satellite will be known in the equatorial coordinate system. In this case the earth's equator is the fundamental plane. Now, the ecliptic and the equatorial coordinate systems have one axis in common. This is the axis formed by the intersection of the equatorial and ecliptic planes. Since the gegenschein coordinate system and the ecliptic system also have an axis in common (the z-axis) then it is a matter of two rotations to transform from the equatorial coordinates to the gegenschein coordinates, or vice versa.

The transformation is carried out in the following manner.

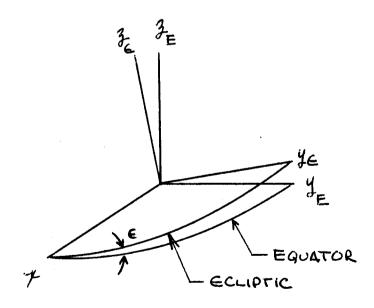


Figure 2

Let x, y_E , z_E = the principal axes in the equatorial system.

x, y_{ϵ} , z_{ϵ} = the principal axes in the ecliptic system.

 ε = the obliquity of the ecliptic.

Consequently, the following relationship between the two systems is written.

$$\begin{bmatrix} \mathbf{x} \\ \mathbf{y}_{\mathbf{E}} \\ \mathbf{z}_{\mathbf{E}} \end{bmatrix} = \begin{bmatrix} \mathbf{1} & \mathbf{0} & \mathbf{0} \\ \mathbf{0} & \cos \varepsilon & -\sin \varepsilon \\ \mathbf{0} & \sin \varepsilon & \cos \varepsilon \end{bmatrix} \begin{bmatrix} \mathbf{x} \\ \mathbf{y}_{\varepsilon} \\ \mathbf{z}_{\varepsilon} \end{bmatrix}$$

In shorthand matrix notation this is written as:

$$C_{E} = T_{I} C_{\epsilon} \tag{1}$$

where C_E and C_e are three-vectors representing the position of the satellite in the equatorial and ecliptic respectively. T is the transformation matrix.

The relationship between the ecliptic and gegenschein systems can also be shown in a sketch.

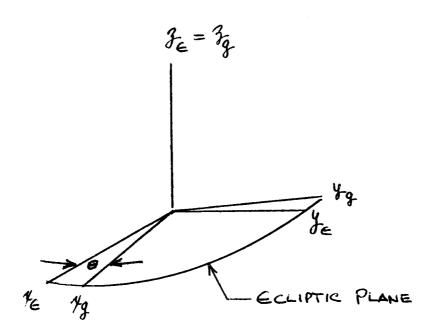


Figure 3

In this case the two systems are related by the following:

$$\begin{bmatrix} x_g \\ y_g \\ z_g \end{bmatrix} = \begin{bmatrix} \cos \theta & \sin \theta & 0 \\ -\sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} x_{\epsilon} \\ y_{\epsilon} \\ z_{\epsilon} \end{bmatrix}$$

or more simply as

$$C_{g} = T_{2} C_{\epsilon}$$
 (2)

where c_g = the position of the satellite in the gegenschein coordinate system.

 T_{2} = the transformation matrix.

Since $^{\rm C}{\rm E}$ is known equation (1) can be solved for C and substituted $^{\rm c}$ in equation (2) to form the following.

$$C_{\varepsilon} = T_1^{-1}C_E = T_1^TC_E$$
 because of orthogonality, therefore

$$C_g = T_2 T_1^T C_E$$

Then
$$C_g = TC_E$$
. (3)

In its expanded form equation (3) appears as:

As a result of equation (4) the position of the satellite is known in the gegenschein coordinate system.

The value for the obliquity, ϵ can be considered a constant. The mean value of the obliquity is given in the American Ephemeris and Nautical Almanac. For 1964, $\epsilon=23.44397$ (Reference 2.) The value for θ can be determined from the position vector of the sun, \overline{r}_s in the ecliptic system.

If
$$\overline{r}_{s_{\epsilon}} = (x, y_{\epsilon}, z_{\epsilon})$$

then $\theta = \tan^{-1} \left(\frac{y_{\epsilon}}{x} \right) \pm \pi$ (5)

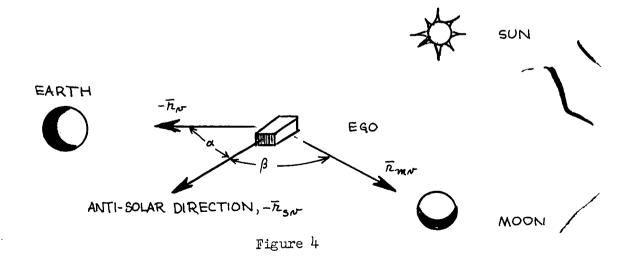
If the first term of (5) is less than π , then π is added; if the first term is greater than π , then π is subtracted. The first term defines the right ascension of the sun along the ecliptic plane and the addition or subtraction of π defines the anti-solar direction.

II. EARTH AND MOON ANGLES

The gegenschein experiment device is quite sensitive to light intensities because of the nature of the planned observations, and it cannot tolerate direct sunlight, earth albedo or moonlight. Sunlight reflected off of the spacecraft itself or off of any of the boommounted experiments may also cause damage to the experiment apparatus. Consequently, it is necessary to know when the sun, earth, moon or any of the boom-mounted experiments will come into the field of view of the gegenschein experiment device. Direct sunlight can be ruled out as a problem because the experiment package is fastened on the dark side of the solar array.

The position of the earth, moon and boom-mounted experiments, however, must be known in order that the gegenschein experiment may be turned off when any of these come into view. The use of a photo-sensitive device is not possible because of weight and reliability considerations.

The earth and moon angles are measured from the satellite and these are defined as the angle between the anti-solar direction and earth or moon respectively. The sketch below illustrates these angles



Since the coordinates of the sun, moon and satellite are known in the earth-equatorial system it is a simple matter of vector subtraction to reference the sun and moon from the satellite.

Let: \bar{r}_v = vector from earth to satellite. \bar{r}_s = vector from earth to sun. \bar{r}_m = vector from earth to moon.

Then the vector from the satellite to the sun or moon is:

$$\overline{r}_{sv} = \overline{r}_{s} - \overline{r}_{v}$$
, vector from satellite to sun $\overline{r}_{mv} = \overline{r}_{m} - \overline{r}_{v}$, vector from satellite to moon.

Unit vectors can be formed by:

$$\overline{U}_{SV} = \frac{\overline{r}_{SV}}{|\overline{r}_{SV}|}$$
, unit vector from satellite to sun.

$$\overline{U}_{mv} = \frac{\overline{r}_{mv}}{|\overline{r}_{mv}|}$$
, unit vector from satellite to moon.

$$\overline{U}_{v} = \frac{\overline{r}_{v}}{|\overline{r}_{v}|}$$
, unit vector from earth to satellite.

Consequently, the earth angle, α and the moon angle, β are:

$$\alpha = \cos^{-1} \{ \overline{U}_{SV} \cdot \overline{U}_{V} \} \text{ where } 0^{O} \le \alpha \le 180^{O}$$
 (6)

$$\beta = \cos^{-1} \left\{ -\overline{U}_{sv} \cdot \overline{U}_{mv} \right\} \text{ where } 0^{\circ} \le \beta \le 180^{\circ}$$
 (7)

The $-\overline{r}_v$ vector points to the center of the earth. As a result it is possible that a portion of the earth can be in the field of view. This will modify β by the angle δ which is measured between a vector to the center of the earth and one to an edge. See the sketch below.

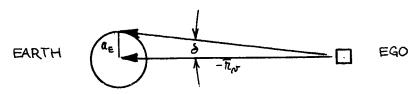


Figure 5

The angle δ is defined by:

$$\delta = \tan^{-1} \left\{ \frac{a_E}{|\overline{r}_v|} \right\}$$
, where $0 \le \delta \le 90^\circ$

Therefore α corrected for the diameter of the earth is:

$$\alpha_{c} = \alpha - \delta \tag{8}$$

III. SPACECRAFT ANGLES

The angles between the gegenschein experiment device and the boom mounted experiments is done in a manner similar to the earth and moon angles. In this case the origin of the reference frame is the location of the gegenschein experiment package.

Since the location of all the experiments are known in the OGO body coordinate system (Reference 3) it is enough to consider these position coordinates as the components of vectors.

If
$$x_i$$
, y_i , z_i where $i = 1, 2, 3, ...$ n

 $n = \text{the number of experiments}$

are the coordinates of the experiments then

$$\overline{\mathbf{r}}_{i} = \overline{\mathbf{i}}\mathbf{x}_{i} + \overline{\mathbf{j}}\mathbf{y}_{i} + \overline{\mathbf{k}}\mathbf{z}_{i} \tag{9}$$

represents the position vectors of the experiments.

Let \overline{r}_g = the position vector of the gegenschien experiment. Then the position of all the experiments can be referenced from the gegenschein experiment by:

$$\overline{r}_{ig} = \overline{r}_{i} - \overline{r}_{g}$$
 (10)

It is necessary to take into consideration that the solar array rotates with respect to the main structure of the satellite. Consequently, the position vector $\overline{\mathbf{r}}_g$ is a function of the solar array angle, φ . The solar array angle, φ is the angle between the normal to the solar cell side of the solar array and the y-axis in the box coordinate system. See figure 6 on page 18.

Let
$$\overline{r}_{go}$$
 = position vector for $m = 0$.

Then the position vector, $r_g(\phi)$ for any ϕ is given by: (Reference 3)

$$\overline{r}_{g}(\varphi) = \begin{bmatrix} x_{g} \\ y_{g} \\ z_{g} \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & x_{g_{0}} \\ 0 & \cos \varphi & \sin \varphi & y_{g_{0}} \\ 0 & \sin \varphi & \cos \varphi & z_{g_{0}} \end{bmatrix}$$
(11)

By substituting (11) into (10) the position vectors from the gegenscheic experiment of all the experiments is determined for all ϕ .

The normal to the gegenschein experiment is defined as:

$$\overline{N}_{g} = (0, -\cos \varphi, \sin \varphi)$$
 (12)

To find the angles, γ between the gegenschein experiment and the ther experiments it is simply necessary to conduct another dot product.

$$\gamma_i = \cos^{-1} \{\overline{U}_{ig} \cdot \overline{N}_g\} \text{ where } 0^0 \le \gamma_i \le 180^0$$
 (13)

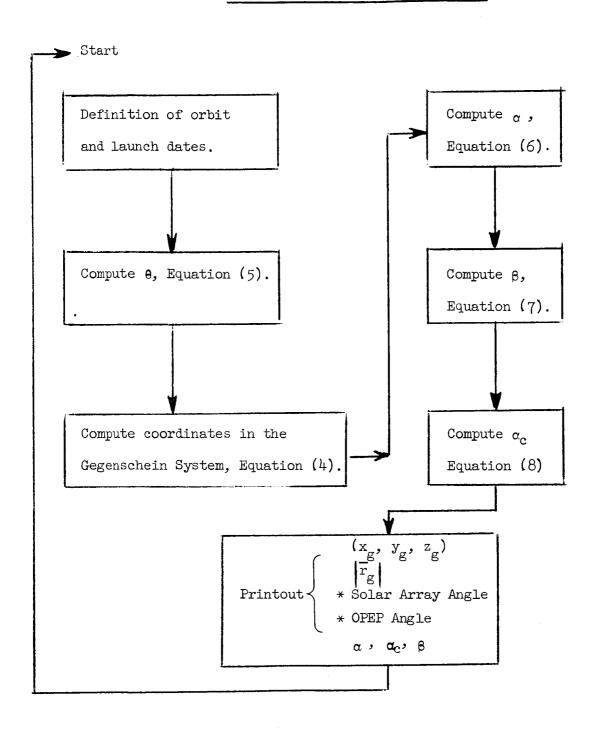
where
$$\overline{U}_{ig} = \frac{\overline{r}_{ig}}{|r_{ig}|}$$

IV. COMPUTER PROGRAMS

Two computer programs are involved with the solutions to the equations presented in preceding sections. One is a modified "Shades of EGO" program (Reference 3) and the other was written specifically to solve these problems. The Shades of EGO program is used to compute the gegenschein coordinate system and the earth and moon angles. The Shades of EGO program also computes the solar array angle, the OPEP angle and the distance of the satellite from the earth.

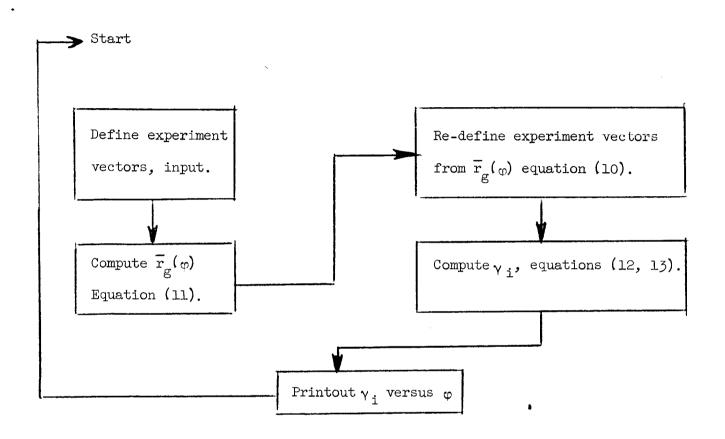
The experiment angles are computed in a separate independent program.

Abbreviated flow charts of the logic for the two programs are shown on the next two pages.



^{*} From main program.

SPACECRAFT ANGLES



The Modified Shades of EGO Program is a numerical integration computer program. It is an adaptation of the ITEM program (Reference 4) which makes use of a modified Encke method of orbit analysis. ITEM is a general purpose interplanetary program. In such a program the orbiting body is "flown" around the orbit in discrete time steps. The Modified Shades of EGO Program is used in conjunction with ephemeris tapes which contain the positions of, among other celestial bodies, the sun and moon. As a result the earth and moon angles are computed. The Shades of EGO Program contains techniques for computing the solar array angle and OPEP angles consequently these are not shown in the flow chart. These, however, are described in Reference 3.

V. THE EGO ORBIT AND EXPERIMENT LOCATIONS

The EGO, S-49 will be placed into a high eccentricity orbit with an apogee about 80,000 nautical miles high. The injection conditions for the EGO are assumed to be the same as the Agena burnout conditions. See Reference 5.

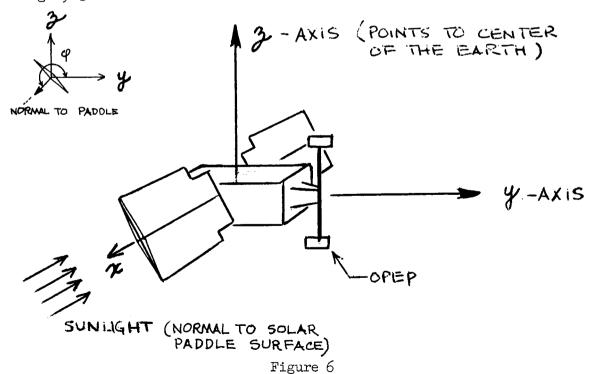
TABLE I

	ON CONDITIONS
Geocentric latitude	20 ⁹ 744573 S
Longitude	111°11923 E
Height	279.2517 km
Speed	10.716286 km per sec
Azimuth	66°445986
Flight Path Angle	1.452111

The classical orbital elements associated with these injection conditions are shown below. These elements are the osculating elements taken at approximately apogee with the exception of the right ascension of the ascending node, Ω which is that of injection.

	EGO ORBITAL ELEMENTS AT FIRST APOGEE
a	12.63422 earth radii
е	.91701983
i	31.020115
ω	-46.341238
Ω	138.07048 (injection)
EPOCH	July 25, 1964 3.0 hrs. U.T.

The location of the boom-mounted experiments and the gegenschein experiment are shown in the Table III below. These are given in the box-coordinate system. In this system the center of gravity of the spacecraft is chosen as the origin. The box coordinate system and the solar array angle, φ are shown in the sketch below.



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TABLE III

	EXPERIMENT (COORDINATES	
Experiment No.	x (inches)	y (inches)	z (inches)
E: P. 1	-74.2	91.1	12.3
E. P. 2	49.9	100.1	15.3
E. P. 3	-12.7	103.1	-4.2
E. P. 4	20.6	-94.4	-19.2
E. P. 5 (edge)	67.2	321.0	-25.2
E. P. 6	-6.7	-288.9	-16.7
OPEP 1	 2	43.1	45.8
OPEP 2	2	43.1	-45.8
Gegenschein Exp.	106.9	-1.3	3.0

In the case of E.P. 5 the coordinates of the edge most likely to come into the field of view of the gegenschein experiment are given. The coordinates of the gegenschein are given for the solar array angle, $\omega = 0$. The coordinates of the gegenschein experiment package for any ω are given by equation (11).

VI. RESULTS AND DISCUSSION

1. Gegenschein Coordinate System

The location of the EGO, S-49 spacecraft in the gegenschein coordinate system as a function of time is determined according to transformations developed in Section I of this document. These data are shown in Appendix A. The launch dates selected for this investigation are chosen in accordance to the launch time restraints imposed on the EGO, S-49. These restraints are discussed in Reference 7. The launch times used for the development of the gegenschein orbital parameters and operation schedule are the following.

Year	Month	Day	Hours (U.T.)
		<u></u> 4	0100
		14	0400
1964	September	9	0100
		9	0330
		14	0100
		14	0300

Results are shown for the first orbit, for an orbit 30 days after launch and finally for an orbit 60 days after launch.

The coordinates in the gegenschein system are a function of flight time and also a function of the date since the position of the sun defines the gegenschein axis system.

2. Earth and Moon Angles

The earth and moon angles (σ_c and β) as functions of time are shown in Appendix B. These were computed for the same launch dates as chosen for the gegenschein coordinate system study (see page 20). Results are shown in Appendix B for the earth and moon angles for the first two orbits, for two orbits 30 days after injection and for two orbits 60 days after injection. If either the earth or the moon come into the field of view of the gegenschein experiment the experiment will have to be turned off. The half angle of the field of view of the experiment is approximately 15°. Consequently, "off" time is indicated for the experiment for earth or moon angles of 15° or less.

The earth and moon angles are functions of flight time and date. The moon angle has a greater daily rate of change than the earth angle simply because the moon is orbiting the earth whereas the earth angle's daily rate of change is a function of the earth's orbital position around the sun.

3. Spacecraft Angles

The spacecraft angles, γ_i are directly related to the solar array angle. Consequently, it is sufficient to know the angles between the normal to the gegenschein and the other experiments solely as a function of the solar array angle. If at a certain solar array angle an experiment package comes into the field of view of the gegenschein experiment then, as before, this indicates an "off" time. The actual solar array angle will be known from telemetry when the satellite is in orbit. Predictions of the solar array angles have been made in the launch window analysis for the EGO orbit, Reference 8.

The computer output for the spacecraft angles is shown in Appendix C. The words "NO GO" are printed in front of the solar array angles for which other experiments come into the field of view of the gegenschein. The specific experiment in view in a "NO GO" case is indicated by an asterisk. In the printout shown in Appendix B the half angle of the field of view was chosen as 15°. The half angle is an input quantity, consequently the "NO GO" solar array angles are functions of the field of view. The Appendix C results show that for the 15° half angle field of view criterion the solar array angles between 170° and 198° have the E.P. 5 (edge) experiment coming into the field of view.

APPENDIX A

ORBITAL POSITION OF EGO
in the
GEGENSCHEIN COORDINATE SYSTEM

TABLES FOR APPENDIX A

Orbital Position of EGO in the Gegenschein Coordinate System

First orbit (Injection time, U. T. = 248^d 1^h 0^m 0^s, 1964). Table A-1 One orbit (Injection time for A-1 + 30 days flight time). Table A-2 One orbit (Injection time for A-1 + 60 days flight time). Table A-3 First orbit (Injection time, U. T. = 248^d 4^h 0^m 0^s, 1964). Table A-4 One orbit (Injection time for A-4 + 30 days flight time). Table A-5 One orbit (Injection time for A-4 + 60 days flight time). Table A-6 First orbit (Injection time, U. T. = 253^d 1^h 0^m 0^s, 1964). Table A-7 One orbit (Injection time for A-7 + 30 days flight time). Table A-8 One orbit (Injection time for A-7 + 60 days flight time). Table A-9 First orbit (Injection time, U. T. = 253^{d} 3^{h} 30^{m} 0^{s} , 1964). Table A-10 Table A-11 One orbit (Injection time for A-10 + 30 days flight time). Table A-12 One orbit (Injection time for A-10 + 60 days flight time). Table A-13 First orbit (Injection time, U. T. = 258^{d} l^h 0^m 0^s, 1964). Table A-14 One orbit (Injection time for A-13 + 30 days flight time). Table A-15 One orbit (Injection time for A-13 + 60 days flight time). Table A-16 First orbit (Injection time, U. T. = 258^{d} 3^{h} 0^{m} 0^{s} , 1964). Table A-17 One orbit (Injection time for A-16 + 30 days flight time). Table A-18 One orbit (Injection time for A-16 + 60 days flight time).

TABLE A-1 X,Y,Z AND R FOR FIRST ORBIT INJECTION AT 1.0 HOURS U.T., SEPTEMBER 4,1964

	01	0	01	01	0	0	0	01	0	01	05	02	02	0	0	02	02	0	02	02	0.2	0	02	02	01	01	0	0	01	0	0	0	0	0	0	0	3 6	7 .	0
α.	.104336	.21865260	,35472858	.47406235	.58046055	.67708301	.76599989	.84861466	.92592460	.99866780	.10674091	.16021989	.19582040	.21952970	.23375473	.23959625	.23745130	.22718024	.20805031	.17839018	.13441826	.11720069	.11089421	.10425631	.97248623	.89823117	.81918158	.73452386	.64314551	.54345865	•43309363	.30851854	.16847000	.12549570	.26273392	512	100000000	10010010.	.70457885
	00	00	01	01	01	01	01	01	0	01	01	02	05	02	05	05	02	05	05	05	01	01	0	01	0	01	01	01	01	01	Ó	00-	00	00	01	010	5 6	7 7	0
2	.70380981	.95655771	.23416424	.34561343	.44051474	.52403358	.59906459	.66741440	. 73029997	.78858811	.84292291	.12421599	.14814706	.16202222	.16817955	.16773020	.16117247	.14858542	.12964196	.10344564	.67952777	.54857779	.50168159	.45295215	.40223052	.34932892	.29402450	.23605423	.17511783	.11091521	.43340815E	.26463236E	.87327059	.23377011	.14130950	0.26980067E	40414016.	.46645382	.54715533
																																				0			
>- ·	.56103641	•69231804	.17634203	0.26315333	.33738969	0.40292815	46196014	0,51586015	0.56555717	0.61171451	0.65482597	0.97478178	0.11712585	0.12899186	0.13485540	0.13553765	0.13137367	0.12235568	0,10812956	0.87835409	.59556068	0,48914909	0.45075367	0.41068662	0.36878625	0.32485694	0.27865891	0.22989477	0.17819415	0.12310639	.64166500	0.14655128	.58480689	.27052850	0.95256913	-0.19458461E	61626112.	349/1528	•41418178
	00	01	01	01	0	0	01	00	00	00-	-01	01	0	01	01	02	02	02	02	02	01	01	01	0	01	01	01	01	01	01	01	01	01	01	01	01	70	01	01
×	0.52773094	.18402687	0.19975726	0,18983360	0.17043892	.14655758	0.12029259	0.92699478	0.64369351	0.35658919	0.67953460	.27181005	.51764968	.72823014	.90393236	.10441497	.11466261	.12067445	.12159726	.11578859	.99517661	.91290519	.88027804	.84445766	.80494522	.76108903	.71200934	.65646943	.59263542	.51758086	.42611539	.30737802	.13166676	0.12029476	.19996714	-0.20920328E	0.19938308	.18158050	0.15970553
TER (HRS)	•	•	•					•		•		0	5	0	5	o	5	ċ	ď.	0	3	•		7	00	ထိ	6	6	o	ċ	-	<u>-</u>	2.	2.	ŝ	63.5	.	4•	5
TIME AFI																																							

TABLE A-2
X,Y,Z AND R FOR ONE ORBIT AFTER 30 DAYS
INJECTION AT 1.0 HOURS U.T., SEPTEMBER 4,1964

	E 02	o C	Ш	E 0	0	E O	<u>п</u>	E 0	E 0	E O	E 0	Е 0	E 0	E 0	E 0	E 0	E 0	E 0	E 0	E 0	E 0	E O	Е 0	Е 0	П 0	E 0	E 0	0	E 0 三	Ε 0	Е 0	П
œ	3048	452	269	367	614	9	754	201	921	015	085	601	666	843	785	008	203	699	553	261	060	138	193	887	329	950	030	960	663	828	135	460
	2373	60	81	39	03	99	92	13	28	37	38	28	04	68	30	62	91	04	07	00	87	89	43	01	08	32	92	90	25	36	38	33
	0	•	•	•	•	•	•	•	•	•	•	•	•	٠	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•
	02) C	0	0	0	0	0	0	0	0	0	0	9	0	9	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	774E	n a	95	83	96	30	46	54	6	22	17	0	33	35	16	59	86	87	33	29	23	65	03	13	64	25	70	54	1	92	8 7	95
7	5906	2007	252	193	751	761	568	154	498	578	814	263	971	139	573	010	938	557	969	780	190	717	842	522	817	395	525	383	357	618	250	284
		• ~		9.	4.	6	6	.2	.2	~	φ,	. 1	0.4	0.9	0.2	7	2	6	4.	3	9	9	7.	8	ω.	7			-4	~	٦.	٦.
	02	200	02	02	01	01	01	01	01	01	01	01	_	Ó	00	00	00	01	01	01	01	10	01	01	01	01	02	02	02	02	05	02
	96E	٥ ر	, W	11	29	72	59	47	28	35	28	73	19	40	55E	82	81	07	64	88	68	63	90	05	57	47	23	75	58	30	23	40
>	1525	101 785	919	340	818	234	721	636	115	940	302	312	980	448	171	645	837	680	961	545	081	120	091	337	136	893	549	620	910	512	856	924
	17	• LO	13	.10	•76	• 70	•64	.58	.51	.43	.34	.25	13	.20	.87	•20	.61	• 13	.21	.27	.34	.40	• 45	.51	• 56	• 75	.11	•13	.15	•16	•17	•17
	Α,	 	, <u>,</u>		1 -	-	-	1	1	-	-	-	1	1	0	-	1	1	-		-	ا ا	1 -	-	<u> </u>	1	1	1	0	1 -	1 -	i ,—i
	6E 0	и ц	נט נ	u	ш	ш	نیا	ш	ш	ш	ш	ننا	ш	ш	ш	ш	ш	لنا	w	ш	ш	ш	ш	نيا	u	نيا	w	ш	Ü	ш	w	ш
×	2569	47	58	58	72	58	64	05	96	41	88	59	9	35	15	67	42	37	93	75	51	32	02	64	89	45	96	20	21	24	50	02
	400	4 7 なんり	622	615	561	545	526	503	476	442	400	345	267	136	43	221	277	307	326	337	343	345	345	343	339	314	212	916	335	156	272	379
	0		0	0	0	0	0	0	0	0	0	0	0	0	1	9	9	î	9	1	9	9	9	9	9	0	0	9	0	0	0	0
TER (HRS)	0	'n	•	o	8	9	4.	4.	5	5	9	•	'	7	œ	œ	6	6	o	Ö	1	-	2.	2	'n	3	ċ	5	•	Š	ċ	Ŝ
AFT ION	+ 5																															
TIME JÉCT	DAY		-				: -		,																				_		>	>
Ž	30																						٠									

TABLE A-3 XXYY,Z AND R FOR ONE ORBIT AFTER 60 DAYS INJECTION AT 1.0 HOURS U.T., SEPTEMBER 4,1964

œ	.12881851	.17433190	.20500866	.22493286	.23588801	.23866650	.23345188	.21989303	.19695784	.16238371	.11065941	.10406005	.97096203	.89721473	.89721473	.81876612	0.73484043E	•64438792	.54593545	.43734817	.31565507	.18077233	.12445931	.25056219	.37957191	•49416994	.59726795	•69145163	.77839270	.85941456	.93538308	.10069717	.10747021	.15598485
7	.10181702E 0	•13349589E 0	.15209451E 0.	.16174135E 0.	.16423801E 0.	.16045979E 0.	•15076947E 0	.13514186E 0	.11313285E 0.	.83646351E 0	.44124455E 0	.39419273E 0	.34539810E 0	.29472397E 0	.29472397E 0	.24201757E 0	0.18711712E 01	.12987580E 0	.70244059E 0	.85655972E-0	.53115164E 0	.10221282E 0	0.38541450E-0	.13058972E 0	.26361773E 0	.37187748E 0	.46446504E 0	.54605172E 0	.61933883E 0	.68604591E 0	.74734657E 0	.80408445E 0	•85689057E 0	.12115790E O
>	.48891515E 0	•86770058E 0	•11631122E 0	0.13906518E 0	0.15572044E 0	0.16648087E 0	0.17116919E 0	0.16918220E 0	0.15926252E 0	0.13881332E 0	0,10130284E 0	0.96002685E 0	0.90278356E 0	0.84062479E 0	0.84062479E 0	0.77265625E 0	-0.69764752E 01	0.61382145E 0	0.51844095E 0	0.40689404E 0	0.27033042E 0	0.89051444E 0	.11718229E 0	.13875525E 0	.98963819E 0	.48339636E-0	0.45995206E-0	.57499835E 0	0.10954132E 0	.16041581E 0	.21001618E 0	.25832187E 0	.30535144E 0	•67788158E 0
×	.61944531E 0	0.71002931E 0	.73263329E 0	0.71382093E 0	0.66484378E 0	0.59147934E 0	0.49693591E 0	0.38301521E 0	.25075872E 0	0.10130056E 0	•60225963E 0	.76223217E 0	.91920054E 0	.10715429E 0	.10715429E 0	.12168721E 0	0.13515026E 01	.14693983E 0	.15598167E 0	.16011650E 0	•15407262E 0	.11958684E 0	0.16522358E-0	.16271053E 0	0.25453229E 0	0.32182969E 0	0.37547314E 0	0.42022038E 0	.45860147E 0	0.49213940E 0	0.52183123E 0	0.54837053E 0	.57226190E 0	0.71111202E 0
TIME AFTER INJECTION (HRS)		-	0	5.		5	•	ح		S	0		-	-		2.	52.5	3		4.		5		9		7	~			6			°	

4,1964 TABLE A-4

X,Y,Z AND R FOR FIRST ORBIT
INJECTION AT 4.0 HOURS U.T., SEPTEMBER

	01	01	01	01	01	02	02	02	02	05	020	02	02	010	01	0 0	01	01	0.0		01		0		
· œ	0.10433647E 0.21865255E 0.35472838E	47406203	.67708306 .76600056	.84861631 .92592761	,99867277 ,10674165	16022678	21957681	,23383996 ,23973435	23765980	.22748089	17897961	.13526799	.10532090	•98 <i>51</i> 0140 •91010523	.83180841	.74803677	.55937845	.45074809	•32845132 •32845132	•18962060 •11163197	.24190604	.37490295	.49173846	.59629530	9149794
7	-0.50484281E 00 0.14579081E 01 0.28120296E 01	.38332350E 0 .46695816E 0	.53848718E 0 .60127804E 0	.65735511E 0	.75427682E 0 .79672477E 0	.10879959E 0	.13037387E 0	.13073142E 0 .12595532E 0	.11659958E 0	.10289583E 0	.62037539E 0	.33812831E 0	.17250671E 0	•13/20026E U •10117892E O	.64494342E 0	.27261835E-0 .10273308F-0	0.47576034E-0	.83372147E 0	0.11403760E 0	.16969515F-0	.17159630E 0	.30010163E 0	.39869027E 0	.48010191E 0	5004825E 0
>	-0.48987430E-01 -0.15040171E 01 -0.21421788E 01	0.25172961E 0 0.27698306E 0	0.29505758E 0 0.30837539E 0	.31828999E 0	0.33470365E 0	0.32280560E 0	0.20059255E 0	•12301578E 0 •41568523E-0	.40914336E-0	•12185775E 0	.196U8/U4E U	.30767091E 0	•30971790E 0	•30/010/3E U •30273863E O	.29650630E 0	.28774301E 0	.25856781E 0	.23406158E 0	.19621629E 0	•12/41823E U	.16075323E 0	.22223162E 0	.26096969E 0	•28839295E 0	0894233E 0
×	-0.91178069E 00 -0.62716033E 00 0.29458846E-00	.12012543E 0	.28533150E 0 .36073008E 0	.43211735E 0	.56468204E 0	.11310720E 0	.17553991E 0	•19349184E 0 •20393759E 0	.20705067E 0	.20251299E 0	.18940648E U .16579178E O	.12730870E 0	•99174870E 0	.92444627E U .85229319E O	.77448676E 0	.68994186E 0	.49374432E 0	.37608257E 0	.23743396E 0	•6458/098E 0	.56847517E 0	.33231641E-0	.12143973E 0	.20469429E 0	8314479E 0
TIME AFTER INJECTION (HRS)	0.0	• •	• •	• •	• •		· •	0	5.	• u	'no	5.	.	. œ	6	60	o	ŗ.	.	, ,	i M	6	.	4•	Ç.

TABLE A-5
X,Y,Z AND R FOR ONE ORBIT AFTER 30 DAYS
INJECTION AT 4.0 HOURS U.T., SEPTEMBER 4,1964

		INJECTION AT 4.	O HOURS U.T., SEP	PIEMBER 4,1964	
. TIME AFI INJECTION	AFTER ON (HRS)	×	>	7	œ
30 DAYS +	0.0	0.18589907E 0	2 -0.94475766E 01	0.11420251E 02	0.23775270E 02
	. 0	•18/31/8/E 0 •18103690E 0	-0.73915078E 0	0.83190668E 0	.21250526E 0
	•	.16533750E 0	-0.57904583E 0	0.61254448E 0	.18558431E 0
	•	.13648634E 0	-0.37165270E 0	0.34514555E 0	•14560574E 0
	ŝ	•10345490E.0	-0.18991808E 0	0.12555733E 0	.10593041E 0
	4.	.97333214E 0	-0.16087066E 0	0.91963173E 0	.99081382E 0
	4•	.90713999E 0	-0.13095523E 0	0.57901891E 0	.91837075E 0
	Š	•83511619E 0	-0.10015067E 0	0.23474717E-0	.84142751E 0
	S.	.75609919E 0	-0.68459564E 0	-0.11136719E-0	•75927380E 0
	•	•66844287E 0	-0.35940614E-0	-0.45597252E-0	•67095953E 0
	9	.56967628E 0	-0.27889584E-0	-0.79251048E 0	.57516915E 0
	۲.	.45578629E 0	0.30433902E-0	-0.11068340E 0	.47001933E 0
	-	•31948406E 0	0.61993298E 0	-0.13629938E 0	.35283247E 0
	7	•31948406E 0	0.61993298E 0	-0.13629938E 0	.35283247E 0
	œ	0.14551182E 0	0.84917221E 0	-0.14376696E 0	.22148039E 0
,	æ	0.78672604E 0	0.53705540E 0	-0.65582685E 0	•11564926E 0
	•	•14496140E 0	-0.75990508E 0	0.13119172E 0	.20976089E 0
	6	0.10354442E 0	-0.17861102E 0	0.27278233E 0	.34210166E 0
	°	0.45787345E-0	-0.25811772E 0	0.37855728E 0	.46046373E 0
	•	.14816438E-0	-0.32413304E 0	0.46440487E 0	.56652769E 0
	-	.75113041E 0	-0.38116577E 0	0.53729632E 0	•66303593E 0
	-	.13413492E 0	-0.43165911E 0	0.60089943E 0	.75193210E 0
	2.	•19156943E 0	-0.47710088E 0	0.65740719E 0	.83457073E 0
	2.	.24733967E 0	-0.51847456E 0	0.70825060E 0	•91192746E 0
	3	•30146116E 0	-0.55646984E 0	0.75442341E 0	.98472949E 0
	3	•35398635E 0	-0.59159285E 0	0.79664905E 0	•10535359E 0
	5.	•50265190E 0	-0.68317484E 0	0.90452883E 0	.12399835E 0
	•	•91899431E 0	-0.89188580E 0	0.11344641E 0	.17108535E 0
	ۍ.	.12411304E 0	-0.10102238E 0	0.12456183E 0	.20279354E 0
	·	•14879086E 0	-0.10690669E 0	0.12807558E 0	.22354220E 0
	5.	•16678260E 0	-0.10812697E 0	0.12581204E 0	.23523722E 0
	0	•17838262E 0	-0.10528813E 0	0.11869920E 0	•23873720E 0
•	٠.	•18346921E 0	-0.98634568E 0	0.10721800E 0	.23427639E 0

TABLE A-6

X,Y,Z AND R FOR ONE ORBIT AFTER 60 DAYS
NJECTION AT 4.0 HOURS U.T., SFPTEMBER 4,19

œ	E 01 0.11357819E 02 E 02 0.16381746E 02 E 02 0.21980263E 02 E 02 0.23282306E 02 E 02 0.23755609E 02 E 02 0.23755609E 02 E 02 0.237431959E 02 E 01 0.22290112E 02 E 01 0.2528376E 01 E 00 0.92603376E 01 E 00 0.92603376E 01 E 00 0.92603376E 01 E 01 0.37689629E 01 E 01 0.25283597E 01 E 01 0.30822642E 01 E 01 0.30822642E 01 E 01 0.42718516E 01 E 01 0.95753390E 01 E 01 0.95753390E 01
7	0.84930193 0.12201363 0.12283293 0.12367363 0.12367363 0.12583293 0.1659469 0.105159469 0.1659469 0.1659469 0.17888238 0.46264820 0.17888238 0.52522953 0.52523841 0.52523841 0.52523841 0.52523841 0.74526059 0.74526059 0.74153900 0.78454018
>	-0.75202571E
×	0.56071453E 00 0.3333331E 01 0.57146540E 01 0.93338222E 01 0.10583039E 02 0.10583039E 02 0.11437666E 02 0.11851728E 02 0.11851728E 02 0.10918789E 02 0.80292237E 01 0.72351424E 01 0.72351424E 01 0.72351424E 01 0.72351426E 01 0.72351426E 01 0.72351426E 01 0.72351426E 01 0.72351426E 01 0.72351426E 01 0.55189652E 01 0.55189652E 01 0.1528952E 01
TIME AFTER JECTION (HRS)	4 +
	AFTER Y Z

TABLE A-7 X,Y,Z AND R FOR FIRST ORBIT INJECTION AT 1.0 HOURS U.T., SEPTEMBER 9,1964

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TIME AFTER INJECTION (HRS)

.10433647E .21865260E .35472869E .47406274E .58046144E .67708457E .76600225E .84861797E .92592888E .92592888E .10674154E .10674154E .10674154E .19581778E .23372377E	0.23734778E 02 0.22701573E 02 0.13883143E 02 0.13383143E 02 0.10348700E 02 0.96429312E 01 0.88946518E 01 0.88946518E 01 0.53100929E 01 0.53100929E 01 0.53109874788E 01 0.5211951E 01 0.15098462E 01 0.15098462E 01 0.29211951E 01 0.28025147E 01 0.28025147E 01 0.28025147E 01 0.28025147E 01
.69142011E 0 .24226920E 0 .24226920E 0 .35415152E 0 .44906297E 0 .53236683E 0 .60704622E 0 .7373364E 0 .73733664E 0 .73733664E 0 .12495766 0 .12415762E 0 .12415762E 0 .12415762E 0 .16537390E 0	0.15841494E 02 0.14552131E 02 0.12638858E 02 0.10016026E 02 0.64877924E 01 0.42482632E 01 0.37483774E 01 0.3276578E 01 0.26840809E 01 0.21153202E 01 0.21153202E 01 0.23770731E-00 0.23770731E-00 0.23770731E-00 0.23770731E-00 0.23770731E-00 0.2975664E 01 0.2935264E 01 0.2935264E 01 0.2935264E 01 0.2935264E 01
0.55755804E 0.67238297E 0.17295681E 0.25878599E 0.33225428E 0.39716023E 0.50909045E 0.55837969E 0.55837969E 0.664696532E 0.64696532E 0.1609610E 0.12797870E 0.13390528E	-0.13063804E 02 -0.12175486E 02 -0.10767161E 02 -0.87516183E 01 -0.59341669E 01 -0.32266183E 01 -0.32266185E 01 -0.22732802E 01 -0.27631275E 01 -0.27732802E 01 -0.17531341E 01 -0.17531341E 01 -0.27631275E 01 -0.37531341E 01 -0.41766581E 01
. 18116743E 0 . 18116743E 0 . 19293528E 0 . 17983881E 0 . 15774992E 0 . 13150220E 0 . 13150220E 0 . 13133776E-0 . 13133776E-0 . 13133776E-0 . 17258957E-0 . 17258957E 0 . 17258957E-0 . 17258957E-0 . 17258957E-0 . 17258957E-0 . 19903837E 0	0.11904780E 0.12464125E 0.12495400E 0.12495400E 0.11831140E 0.10089733E 0.85058028E 0.85058028E 0.76345264E 0.76345264E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.71224996E 0.712996E 0.71299
0011100004440000000	

TABLE A-8

X.Y.Z AND R FOR ONE ORBIT AFTER 30 DAYS
INJECTION AT 1.0 HOURS U.I., SEPTEMBER 9,1964

	000	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	Ö	ö	ö	ö	0	õ	0	0
α	0.23684252E	.20780717	.17835342	.13466465	.10472941	.97782913	.90427306	.82603788	.74235252	.65217250	.55402783	.44576740	.32427891	.18806074	.11747931	.24311224	.37405448	•48988088	.59384207	.68862914	•77609608	.85752118	.93382706	.10057041	.10736852	.13674620	.17974726	.20873595	.22731063	.23707254	.23871543	.23234795
	02																															
7	0.15548775E	12311117	.96953662	.62097380	.40191776	.35332187	.30284325	.25032470	.19559547	.13849115	.78923064	.17132045	.45186696	.97368234	.45083401	.12723878	.26213124	.37121372	.46421877	.54603114	.61944047	.68620896	.74753332	.80427057	.85706008	.10759424	.13680825	.15385685	.16233880	.16386941	.15923180	.14872868
	02																															
≻	-0.17296133E	0.15593144	0.13507808	0.10195936	0.78148534	0.72496157	0.66455345	0.59963936	0.52938486	0.45262344	0.36763519	0.27170040	0.16014734	0.25325718	0.87957047	0.31381613	0.51952479	0,13123625	0.20503315	0.27394655	0.33866884	0.39977352	0.45771541	0.51285583	0.56548575	0.79803619	0.11544005	0.14101194	0.15886616	0.17003159	.17489830	0.17341094
	010																															
×	0.44747245E	60933103	.64527850	.62311905	.56970865	.55294141	.53325835	.51000862	.48226689	.44864373	.40689354	.35297869	.27833039	.15888637	0.63501454	0.20476584	0.26173442	0.29149702	0.30840033	.31782603	0.32236941	0.32348259	0.32206139	0.31869825	0.31380553	0.27466090	0.16314837	0.37946957	.88191424	.20963429	.32262634	.42355630
ER (HRS)	0.0		5.	•	2.	3	3.	4.	4•	•	5.	•	• 9	7.	~	φ	• ф	6	6	•	o	-	-	2.	2.	5.	ं	5	o	5	o	5.
AFTEI ION (I	+																													-		
I ME E C T	DAY!			-,				, , .						·····															,		>	-
LNI	30																															

TABLE A-9

X,Y,Z AND R FOR ONE ORBIT AFTER 60 DAYS
INJECTION AT 1.0 HOURS U.T., SEPTEMBER 9,1964

œ	0.11922625E 02 0.20006079E 02 0.22142490E 02 0.23364206E 02 0.23360913E 02 0.23360913E 02 0.23360913E 02 0.19995528E 02 0.19995528E 02 0.19995528E 02 0.19995528E 02 0.16740149E 02 0.19995528E 02 0.16740149E 01 0.22135758E 01 0.95605752E 01 0.86999050E 01 0.76981005E 01 0.886338513E 01 0.375052068E 01 0.375052068E 01 0.375052068E 01 0.37502635E 01 0.43012007E 01 0.63509452E 01 0.63509452E 01 0.72502635E 01
7	0.88933453E 01 0.12524856E 02 0.12524856E 02 0.125908541E 02 0.12692611E 02 0.12692611E 02 0.10814270E 02 0.2268313E 01 0.72135159E 01 0.72135159E 01 0.72135159E 01 0.72135159E 01 0.17791448E 01 0.17791448E 01 0.23997396E-00 -0.23997396E-00 -0.23997396E-00 -0.23997396E-00 0.12185817E 01 -0.15632465E 01 0.23621844E 01 0.23621844E 01 0.23621844E 01 0.2362465E 01 0.2362465E 01 0.2362465E 01 0.2362465E 01 0.2362465E 01 0.2362466E 01 0.2362466E 01 0.2362466E 01 0.2362466E 01 0.2362466E 01 0.2362466E 01 0.2362466E 01 0.2362466E 01 0.2362466E 01
>	-0.79328727E 01 -0.12197914E 02 -0.15137640E 02 -0.17134536E 02 -0.18855063E 02 -0.18855063E 02 -0.18863820E 02 -0.18863820E 02 -0.18863820E 02 -0.18863820E 02 -0.13249809E 02 -0.13744253E 01 -0.52452739E 01 -0.32084206E-00 -0.32084206E-00 -0.32084206E-00 -0.32084206E-00 -0.32084206E-00 -0.32084206E-00 -0.32084206E-00 -0.32084206E-00
×	-0.35630535E-00 0.37713427E 01 0.54833461E 01 0.69317741E 01 0.89691505E 01 0.94867743E 01 0.95724607E 01 0.95724607E 01 0.95724607E 01 0.95724607E 01 0.95724607E 01 0.95724607E 01 0.95724607E 01 0.95724607E 01 0.95724607E 01 0.19520408E 01 0.19520408E 01 0.19520408E 01 0.19520408E 01 0.19520408E 01 0.21213952E 01 -0.19969870E 01 -0.19969870E 01 -0.19969870E 01 -0.19969870E 01 -0.19969870E 01 -0.19969870E 01 -0.19969870E 01 -0.15711146E 01 -0.12068226E 01 -0.12068226E 01
TIME AFTER INJECTION (HRS)	4

TABLE A-10
X,Y,Z AND R FOR FIRST ORBIT
JECTION AT 3.5 HOURS U.I., SEPTEMBER 9,1964

œ

7

Z AND R FOR FIRS	>
X,Y,Z NJECTION AT 3.5	×
e i	TIME AFTER INJECTION (HRS)

0000000	001	000000	001000000000000000000000000000000000000	000000000000000000000000000000000000000
0433647 1865257 5472850 7406221 8046040 7708288	.84861476 .92592490 .99866844 .10674103 .16022138	.21954208 .23377824 .23963561 .23751199 .22726882	856428 466734 456654 575194 168200 284578 843928	
.52052145E 0 .14379900E 0 .28037057E 0 .38378937E 0 .46870709E 0 .54147616E 0	.66268288E 0 .71447659E 0 .76177000E 0 .80524298E 0 .11051435E 0	.13322219E 0 .13392798E 0 .12937076E 0 .12010403E 0 .10635015E 0	868938E 0 935375E 0 837563E 0 177406E 0 436161E 0 177438E 0 322573E-0	.41173580E .78961964E .11138280E .12087765E .35938082E .30996478E .40755561E .48879719E
.67842925E-0 .14189299E 0 .21729669E 0 .26628620E 0 .30235820E 0	0.37253226E 0.38853277E 0.40212824E 0.41374841E 0.46630456E 0.45980359E	0.42383809E 0.36909555E 0.30081920E 0.22209010E 0.13508067E 0.41830468E	.54693373E 0 14729890E 0 .18500173E 0 .19077754E 0 .19557331E 0 .19508267E 0 .20085060E 0	0.19578332E 01 0.18534947E 01 0.16329498E 01 0.11094577E 01 -0.39218156E-00 -0.23312959E 01 -0.28165272E 01 -0.34884168E 01
.90170101E 0 .83649115E 0 .26079874E 0 .80823030E 0 .16070925E 0 .23652724E 0	.37711821E 0 .44260868E 0 .50530983E 0 .56547386E 0 .10622180E 0	16927547E 0 18802459E 0 19945805E 0 20370010E 0 20039535E 0	16627481E 0 12894561E 0 10117832E 0 94480708E 0 87275595E 0 79475676E 0	5102341 3891186 2442870 5907885 1143823 7222066 7811673 8836283 1656143
• • • • • •	W 4 4 W O W	000000	00/00/00/00	6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6

TABLE A-11 X,Y,Z AND R FOR ONE ORBIT AFTER 30 DAYS INJECTION AT 3.5 HOURS U.T., SEPTEMBER 9,1964

	16E 0 95E 0	59E	89E	35E	14E	01E	72E	50E	38E	36E	16E	57E	40E	24E	38E	98E	67E	55E	57E	67E	83E	84E	19E	13E	50E	88E	14E	72E	06E	34E	84E	32E
œ	37385 28661	127	376	294	227	223	747	785	256	046	966	884	414	141	945	191	689	820	883	105	645	919	100	216	884	701	307	412	435	557	862	369
	0.23	.2	4	• 1	•1	6	œ		7.	•	₹,	4.	•2		٦.	•2	• 4	3	•	7.		8	6•	-	7	• 1	۲.	.2	•2	•2	•2	•2
	02	0	0	0	0	0	0	9	9	0	Ó	0	0	0	9.	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
7	4969E 6583E	497	809	371	788	241	323	653	892	217	332	020	940	367	545	515	170	323	437	889	043	251	291	561	109	660	354	308	885	696	543	315
,7	11744	593	343	580	291	395	820	204	431	044	544	171	391	226	050	089	294	246	045	730	339	884	378	829	244	311	161	275	312	291	220	104
	00	•	•	•	•	•	•	•	0	•	•	•	•	•		•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•
	E 02	0	0	0	0	0	0	0	0	0	9	0	9	0	9	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
	2065E 3108E	092	262	915	550	806	442	153	99	617	414	243	640	445	108	344	470	555	008	789	356	265	680	914	966	567	561	683	200	145	364	090
	163 078	481	999	210	957	585	198	795	373	320	710	387	784	914	083	176	122	106	586	189	734	233	693	120	519	577	600	162	247	279	262	200
	-0.1	•	•	°	o	ਂ	•	·	•	•	•	•	•	•	•	•	0	•	•	o	•	•	•	o	Ö	o	o	0	•	•	ô	°
	02																															
	251E 123E	66	29	63	84	64	31	55	9	32	52	96	85	78	07	59	90	33	11	74	52	48	77	62	47	43	92	05	12	05	70	89
×	7037	811	644	820	053	163	748	708	905	134	075	170	307	291	392	816	917	851	108	892	322	082	926	658	276	470	196	907	242	986	157	735
	0.1	-		~	6	6	φ.	7	9	9	5	6	.2	4.	0.1	0.1	۲.	9.0	0.1	7.	6.	~	۲.	.2	.2	4.	7.	۲,	٦,	۲.		۳,
RS)	0.0	•	•	•	•	•	•	•	•	•	•	•	•	•	ď	0	r.	0	5	•	•	•	•	•	•	•	•	•	•	•	•	•
TER (HR	. .								2																							
E AF TION	YS +																															
TIM JEC	DA -													-															•		_	_
Z	30																															

X,Y,Z AND R FOR ONE ORBIT AFTER 60 DAYS INJECTION AT 3.5 HOURS U.I., SEPTEMBER 9,1964 TABLE A-12

	000000000000000000000000000000000000000
œ	0.13850870E 0.18083641E 0.20936568E 0.22757117E 0.23833899E 0.231650521E 0.21650521E 0.19168829E 0.19168829E 0.19168829E 0.19169829E 0.91193927E 0.91193927E 0.91193927E 0.91193927E 0.956562322E 0.91193927E 0.26474473E 0.26474473E 0.26469085E 0.26474473E 0.26474473E 0.26474473E 0.26474473E 0.26474473E 0.26474473E 0.26474473E 0.26474473E 0.26474473E 0.26474473E 0.26474473E 0.26474473E 0.264769085E 0.264769085E
	0052 0052 001 001 001 001 001 001 001
7	0.10907388E 0.13712776E 0.15321666E 0.16085245E 0.16162372E 0.15630526E 0.12630526E 0.12630526E 0.12630526E 0.12630526E 0.12630526E 0.12630526E 0.22899148E 0.27992080E 0.27992080E 0.27992080E 0.27992080E 0.27992080E 0.27992080E 0.27992080E 0.27992080E 0.27899148E 0.17610183E 0.27899148E 0.27899148E 0.27899148E 0.27899148E 0.27899148E 0.27899148E 0.27899148E 0.27899148E 0.2789914803E 0.25131617E 0.25131617E 0.36387253E 0.45898690E 0.36375739E 0.45898690E 0.45898690E 0.45898690E 0.45898690E 0.45898690E
	000000000000000000000000000000000000000
> -	-0.58812281E -0.12421586E -0.12421586E -0.14602742E -0.16166729E -0.17128247E -0.17101093E -0.13550956E -0.98168328E -0.92425192E -0.98168328E -0.92425192E -0.98168328E -0.92425192E -0.13550956E -0.92425192E -0.129828E -0.129828E -0.129828E -0.129828E -0.129828E -0.129828E -0.129828E -0.129828E -0.30006997E
×	-0.61876211E -0.70207286E -0.70207286E -0.67757827E -0.55104082E -0.55104082E -0.34414132E -0.34414132E -0.21423361E -0.21423361E -0.21423361E -0.1398885E 0.11521631E 0.1398885E 0.15083789E 0.15083789E 0.15083789E 0.15083789E 0.15083789E -0.29935716E -0.29935716E -0.29935716E -0.29935716E -0.29935716E -0.29935716E
ER (HRS)	00000000000000000000000000000000000000
TIME AFTE	+ SAM + + SA

R FOR FIRST ORBIT TABLE A-13 AND X,Y,Z

0.58046073E 0.76600109E 0.97065441E 0.89623100E 0.81698301E 0.73208597E 0.54033750E 0.26791984E 0.10433647E 0.21865255E 0.35472849E 0.47406229E 0.67708360E 0.84861668E 0.92592772E 0.99867229E 0.10674151E 0.16022309E 0.19582707E 0.21953917E 0.23376501E 0.23960414E 0.23745258E 0.22716948E 0.13430127E 0.10408766E 0.64041000E 0.42944419E 0.30409750E 0.16326716E 0.12885081E 0.39778083E 0.51209959E 0.61476110E 847887E 0.20802020E 0.17832904E α 0.708 01 01 01 02 02 02 02 02 00 01 01 01 0.14137698E-00 -0.48383838E-00 01 -0.76293735E-01 AT 1.0 HOURS U.I., SEPTEMBER 14,1964 -0.97056480E 0.10877321E 0.24975900E 0.36170971E 0.45631301E 0.53911696E 0.61318528E 0.68041320E 0.74206959E 0.79905391E 0.85203207E 0.12368000E 0.14620629E 0.15880083E 0.16382623E 0.16241176E 0.15507849E 0.14193988E 0.12272562E 0.96643190E 0.61879611E 0.40002687E 0.35143493E 0.24833912E 0.19348270E 0.13617621E 0.76311327E 0.16295034E 0.29210407E 0.39729226E 6698203E .67641104E 0.30092825E 0.48743007E 'n 0.23587410E-00 -0.57388756E-01 0.55926678E -0.69534038E -0.34312682E -0.40644587E -0.12911312E -0.94860742E .55636337E -0.65461830E -0.17019439E -0.25537606E -0.32836502E -0.39289485E -0.45108491E -0.50426792E -0.55334614E -0.59896456E -0.64160380E -0.95909524E -0.11553641E -0.12750483E -0.13355321E -0.13448047E -0.13060668E -0.12191185E -0.10802896E -0.88079316E -0.60105012E -0.41718978E -0.37539931E -0.33153057E -0.28532958E -0.23647825E -0.18457404E -0,19138356E -0.27238763E 01 00 00 02 02 02 0.10032237E-00 .41964473E-00 01 01 -0.21975874E-00 01 01 01 0 01 0 01 01 0 01 0 -0.14453203E .34293741E .60193826E .81989446E •99852797E 0.12359727E 0.72413809E 0.59792428E 0.51910597E -0.17802269E -0.18570571E -0.16936416E -0.11588243E -0.85345744E -0.53870465E 0.11378737E .12882758E 0.12861014E 0.12125748E 0.10293267E 0.82324895E 0.77637599E 0.66527615E 0.42354158E 0.30016889E 0.11877883E -0.12644350E -0.19034092E -C.19046557E -0.56704314E 0.86565311E -0.17380977E 5035667 359387 INJECTION × -0.1 -0.1 0 40.0 51.5 61.5 62.0 0.0 0.5 1.0 1.5 2.0 2.5 3.0 3.5 0.4 \$.0 5.0 0.0 5.0 0.0 5.0 0.0 5.0 45.0 0.0 5.0 58.0 58.5 59.0 59.5 0.09 60.5 61.0 62.5 63.0 63.5 S INJECTION (HRS) 94•(4 TIME AFTER

02 02 02 02 02

01

02 02

TABLE A-14 X,Y,Z AND R FOR ONE ORBIT AFTER 30 DAYS INJECTION AT 1.0 HOURS U.T., SEPTEMBER 14,1964

c c '	0.23703266E 02 0.22721221E 02 0.17954382E 02 0.17954382E 02 0.13648483E 02 0.13648483E 02 0.10706804E 02 0.10706804E 02 0.93061808E 01 0.93061808E 01 0.77263187E 01 0.48587332E 01 0.36980156E 01 0.36980156E 01 0.36980156E 01 0.36980156E 01 0.36980156E 01 0.36980156E 01 0.36980156E 01 0.36980156E 01 0.36980156E 01 0.3770234E 01 0.32770234E 01 0.32770234E 01 0.32770234E 01 0.32770234E 01 0.32770234E 01 0.32770234E 01 0.32770234E 01 0.32770234E 02 0.23867710E 02 0.23667229E 02
7	0.15232810E 02 0.13906883E 02 0.94002778E 01 0.59889981E 01 0.38673346E 01 0.38673346E 01 0.29149806E 01 0.29149806E 01 0.29149806E 01 0.24123929E 01 0.2415103E-00 0.2693608E 00 0.22382573E 01 0.34140350E 01 0.34140350E 01 0.52489146E 01 0.52489146E 01 0.5248916E 01 0.5248916E 01 0.5248916E 01 0.5248916E 01 0.5248916E 01 0.5248916E 01 0.5248916E 02 0.15599348E 02
>	-0.17508237E 02 -0.17035149E 02 -0.13766177E 02 -0.10469628E 02 -0.81056067E 01 -0.81056067E 01 -0.75460082E 01 -0.40366758E 01 -0.40366758E 01 -0.40366758E 01 -0.40366758E 01 -0.40366758E 01 -0.40366758E 01 -0.20432591E 01 -0.31088265E 01 -0.40366758E 01 -0.40366758E 01 -0.20432591E 01 -0.31088265E 01 -0.4036675E 01 -0.316889060E 01 -0.31580074E 01 -0.31580074E 01 -0.31580074E 01 -0.31580074E 01 -0.31580074E 01 -0.31580074E 01 -0.31580074E 01 -0.316009041E 02 -0.17689461E 02
×	0.48236901E 01.0.5374556E 01.0.63654360E 01.0.65698586E 01.0.658291107E 01.0.58291107E 01.0.5658396E 01.0.56293788E 01.0.56293788E 01.0.5629378E 01.0.5629396E 01.0.29350653E 01.0.29350659E 01.0.29350659E 01.0.29350659E 01.0.2935065E 01.0.293506578E 01.0.2568434E 01.0.24568434E 01.0.246
TIME AFTER INJECTION (HRS)	30 DAYS + 10 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0

TABLE A-15
X*Y*Z AND R FOR ONE ORBIT AFTER 60 DAYS
INJECTION AT 1.0 HOURS U.T., SEPTEMBER 14:1964

œ	FE 02 0.13547505E 02 78 4466E 02 0.17870542E 02 0.17870542E 02 0.20784466E 02 0.20784466E 02 0.20784466E 02 0.22653386E 02 0.23639105E 02 0.23639105E 02 0.238119900E 02 0.238119900E 02 0.23183300E 02 0.2318341E 01 0.29549092E 01 0.29549092E 01 0.29549092E 01 0.29549092E 01 0.2954959E 01 0.295495336E 01 0.2954965236E 01 0.295495336E 01 0.295495336E 01 0.2954965236E 01 0.2954966E 01 0.2954965246E
7	01 0.10708465 02 0.13518704 02 0.15103767 02 0.15836456 02 0.15836456 02 0.15837856 02 0.15327898 02 0.1673222 03 0.10196425 01 0.27064168 01 0.29921005 00 0.29921005 00 0.29921005 00 0.29921005 01 0.29921005 01 0.29921005 01 0.29921005 01 0.29921005
>	-0.58013198E -0.12560742E -0.12560742E -0.16425816E -0.17423478E -0.17423478E -0.17431111E -0.17431111E -0.16425816E -0.17423478E -0.16425816E -0.16425816E -0.16423478E -0.16423478E -0.1643365E -0.16423475377E -0.164348798E -0.269976875 -0.2699776875 -0.269976875 -0.269976875 -0.269976875 -0.269976875 -0.269976875 -0.269976875 -0.269976875 -0.269976875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.2699776875 -0.269977
×	-0.59336618E 01 -0.66540030E 01 -0.65591815E 01 -0.65591815E 01 -0.65591815E 01 -0.53370916E 01 -0.33306997E 01 -0.33306997E 01 -0.44226380E 01 -0.44226380E 01 -0.44226380E 01 -0.44226380E 01 -0.46591815E 01
TIME AFTER INJECTION (HRS)	4 +

TABLE A-16

X,Y,Z AND R FOR FIRST ORBIT NJECTION AT 3.0 HOURS U.T., SEPTEMBER 14,196

0 0.13021019E 0.26983623E 0.39954331E 0.51371947E 0.61626790E 0.70989461E 0.21865254E 0.47406225E 0.67708352E 0.84861656E 0.99867199E 0.19582615E 0.21953745E 0.23376231E 0.23960022E 0.23744681E 0.22716053E 0.20800547E 0.17830370E 0.13425508E 0.10402186E 0.96994148E 0.89545471E 0.81613219E 0.73114578E 0.63935903E 0.53914401E 0.42805612E 0.30242856E 0.16134534E 0.35472846E 0.58046067E 0.76600099E 0.92592749E 0.10674147E 0.16022277E 0.10433647 -0.33944108E-00 01 0 02 0 0 0 0.48205878E-00 0 0.21887357E-00 01 0 01 0 0 0.71702903E-01 INJECTION AT 3.0 HOURS U.T., SEPTEMBER 14,1964 0.88649743E -0.73799231E -0.10818367E -0.11682552E -0.53582498E 0.14167208E 0.27928987E 0.38392320E 0.47005630E 0.54400806E 0.60913402E 0.66745941E 0.72031632E 0.76863639E 0.81310160E 0.11214174E 0.12842784E 0.13597186E 0.13702936E 0.13269921E 0.10974792E 0.91270113E 0.67698950E 0.38098376E 0.20489609E 0.16704838E 0.12829155E 0.19747693E 0.32059418E 0.41751269E 0.49871188E 0.56914155E 0.12353423E 00 0.18192708E-00 00 00 00 00 0 01 01 01 01 01 01 0 0 01 01 01 01 0 01 01 01 01 0 -0.18804419E-00 0 0 0 01 01 -0.39806029E-00 0 01 0 0 0 -0.13058149E -0.32125515E -0.59870867E -0.55265833E -0.15805227E 0.53906581E -0.16138044E -0.21589583E -0.27522781E -0.35897899E -0.39091575E -0.41852930E -0.44275570E -0.46423489E -0.48342645E -0.63830825E -0.47994265E -0.38947484E -0.28205357E 0.82073997E 0.10802735E 0.11924259E 0.12818518E 0.13293935E 0.12860654E 0.96330696E -0.23771918E -0.29447447E -0,34014621E -0.37850523E -0.63649257E -0.60591626E 0.68154842E 0.95467372E 0 -0.34906169E-00 0.39845276E-00 02 02 02 02 02 02 02 -0.18542334E-00 01 01 01 01 01 0 02 01 0 01 0 -0.87658662E -0.10337518E 0.19168952E 0.25143525E 0.55718157E -0,12202917E -0.88152200E 0.53624979E 0.12398855E 0.11303446E 0.18338870E 0.25079506E 0.31540646E 0.37743016E 0.43706959E 0.49450867E 0.97524582E 0.16017839E 0.17943363E 0.19701988E 0.19503947E 0.18477148E 0.16419275E 0.12872220E 0.10184138E 0.95301430E 0.88240823E 0.80566677E 0.62810013E 0.52258265E 0.40014104E 0.19173433E 0.13334090E 0,72151261 0.0 61.5 62.0 35.0 55.0 60.5 62.5 63.0 0.5 1.0 1.5 2.0 3.0 3.5 4.0 4.5 5.0 0.0 15.0 20.0 25.0 30.0 45.0 50.0 58.0 58.5 59.0 59.5 0.09 61.0 63.5 0.49 64.5 65.0 2.5 INJECTION (HRS) TIME AFTER

TABLE A-17

X,Y,Z AND R FOR ONE ORBIT AFTER 30 DAYS INJECTION AT 3.0 HOURS U.T., SEPTEMBER 14,1964

ME AF	S	× .	>- (c)	2	α
DAYS +	0.0	0.15246139E 02 0.15569522E 02	-0.13593006E 02 -0.12751336E 02	0.12013749E 02 0.10621716E 02	0.23696935E 02 0.2275582E 02
		15236787E 0 14075483E 0	-0.11367858E 0 -0.93561295E 0	.87780814E 0 .64547664E 0	.20939021E 0 .18091998E 0
 -		11701521E 0	-0.65265250E 0	.35833386E 0	.13869443E 0
	8	.92855373E 0	-0.42562893E 0	.15490599E 0	.10331349E 0
	რ :	87624725E 0	-0.38180024E 0	.11853401E 0	.96313600E 0 .88898023F 0
	• •	.75610000E 0	-0.28737593E 0	.43905640E-0	.81006162E 0
	5	.68603567E 0	-0.23620552E 0	0.59084415E-0	•72558451E 0
	ري م	.60698653E 0	-0.18189708E 0	.32136218E-U	.63446980E U
	9	.40790284E 0	-0.62000559E 0	0.10435336E 0	.42558010E 0
	7	.27251308E 0	0.37664767E-0	0.13159921E 0	.30264817E 0
	7	.85576548E 0	0.65392093E 0	0.12907353E 0	.16810547E 0
- • • •	• Ф	.12840624E 0	0.36278282E-0	.10326946E-0	•13383169E 0
		.16493027E U	-0.18187089F O	.188/2001E U	.39094543F 0
<u> </u>	• 6	0.10706279E 0	-0.26672869E 0	.41420094E 0	.50415181E 0
	0	0.66860109E 0	-0.34098610E 0	.49672437E 0	.60619872E 0
	•	.25094122E-0	-0.40756957E 0	.56797603E 0	•69952800E 0
	•	.17038025E-0	-0.46821091E 0	.63088305E 0	.78582770E 0
	.	.58974718E 0	-0.52405307E 0	.68/2/142E U	•86628611E U
	• v c	• 10045105E 0	-0.42432589E O	78504796F O	.10129350F 0
	1 (.18138123F 0	-0.66978720F 0	.82795446E 0	.10802881E 0
	r)	.33430510E 0	-0.82795399E 0	.97032544E 0	.13186353E 0
	0	.66608080E 0	-0.11121174E 0	.11953172E 0	.17633071E 0
	5	.93530804E 0	-0.12938064E 0	.13068410E 0	.20631456E 0
	0	.11501925E 0	-0.14028048E 0	.13427598E 0	.22569466E 0
	5.	.13145301E 0	-0.14519844E 0	.13198216E 0	.23618165E 0
>	0	.14288925E 0	-0.14470831E 0	.12465689E 0	.23853128E 0
	5	.14908759E 0	-0.13895790E 0	.11272476E 0	•23290186E 0

TABLE A-18

X,Y,Z AND R FOR ONE ORBIT AFTER 60 DAYS INJECTION AT 3.0 HOURS U.T., SEPTEMBER 14,1964

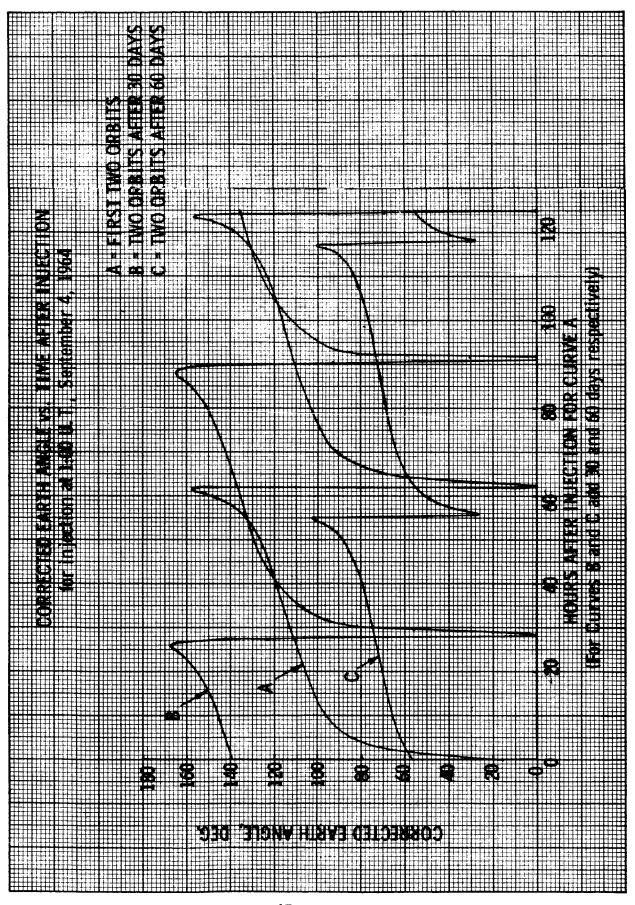
	000000000000000000000000000000000000000
œ	0.12940631E 0.22428597E 0.23498877E 0.23753787E 0.23753787E 0.23753787E 0.23753787E 0.23753787E 0.19516337E 0.19516357E 0.19516355E 0.19516355E 0.19516355E 0.19516355E 0.2782642E 0.218329966E 0.27826127E 0.21831979E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.28377036E 0.27826127E 0.27826127E 0.27826127E 0.27826127E 0.28377036E
7	0.95901992E 01 0.12908881E 02 0.13220372E 02 0.13220372E 02 0.12941137E 02 0.12160856E 02 0.12160856E 02 0.12160856E 02 0.12160856E 02 0.12160856E 01 0.1324687E 01 0.13841312E 01 0.13841312E 01 0.13841312E 01 0.13841312E 01 0.13841312E 01 0.13841312E 01 0.13841312E 01 0.25316844E 01 0.25316844E 01 0.25316844E 01 0.25316844E 01 0.25316844E 01 0.26342288E-01 0.33130492E-00 0.26342288E-01 0.26342288E-01 0.26342288E-01 0.26342288E-01 0.26342288E-01 0.2634238E-01 0.29248194E-00 0.29248194E-00 0.29248194E-01 0.29248194E-01 0.29248169E 01 0.29248169E 01 0.57549047E 01 0.6379588169E 01 0.6379588169E 01 0.63795830E 01
>	-0.86112983E 01 -0.15760831E 02 -0.17806338E 02 -0.19070759E 02 -0.19608980E 02 -0.19623190E 02 -0.18465493E 02 -0.18465493E 02 -0.18465493E 02 -0.18465493E 02 -0.18465493E 02 -0.18465493E 02 -0.18465493E 01 -0.23284769E 01 -0.45612503E 01 -0.45672764E 00 -0.23209108E 01 -0.23209108E 01 -0.23209108E 01 -0.23209108E 01 -0.23214026E 00 -0.23214026E 01 -0.23314026E 01 -0.23312512E 01 -0.24470711E 01 -0.25277867E 01 -0.265589017E 01 -0.59576277E 01
×	-0.11548002E 01 0.42149007E-00 0.19479925E 01 0.45858871E 01 0.56430389E 01 0.70786619E 01 0.7078619E 01 0.70860628E 01
TIME AFTER INJECTION (HRS)	4 +

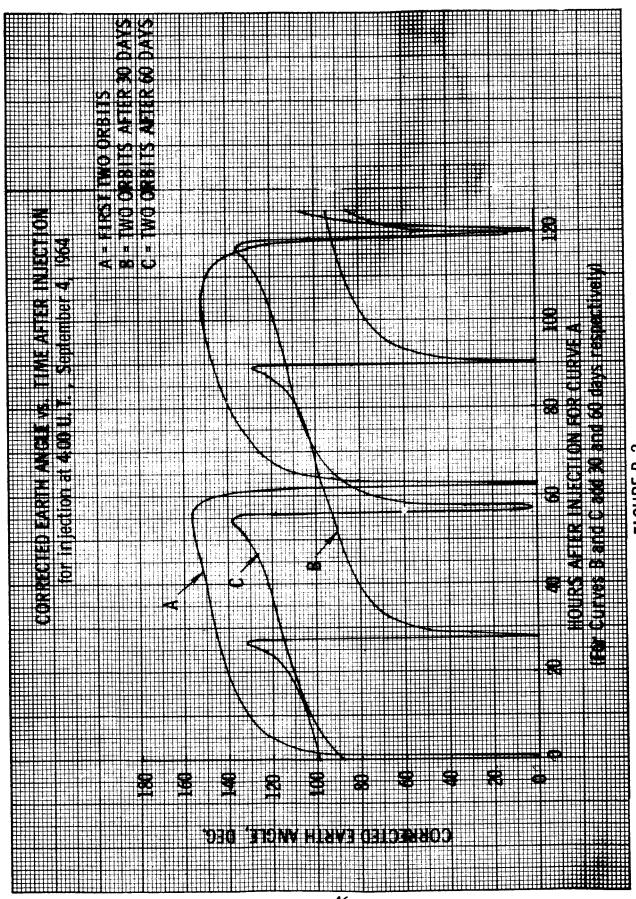
APPENDIX B EARTH AND MOON ANGLES

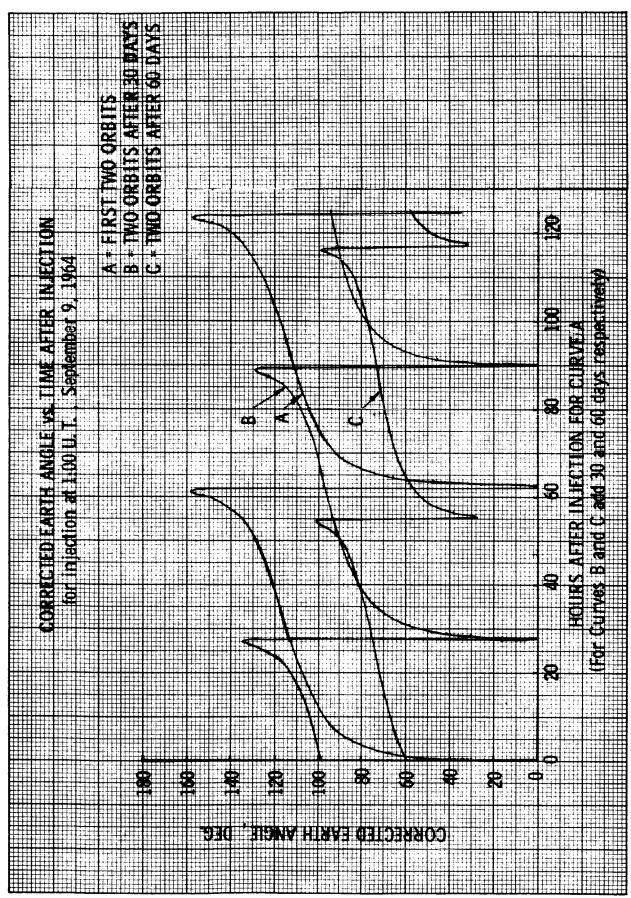
LIST OF FIGURES FOR APPENDIX B

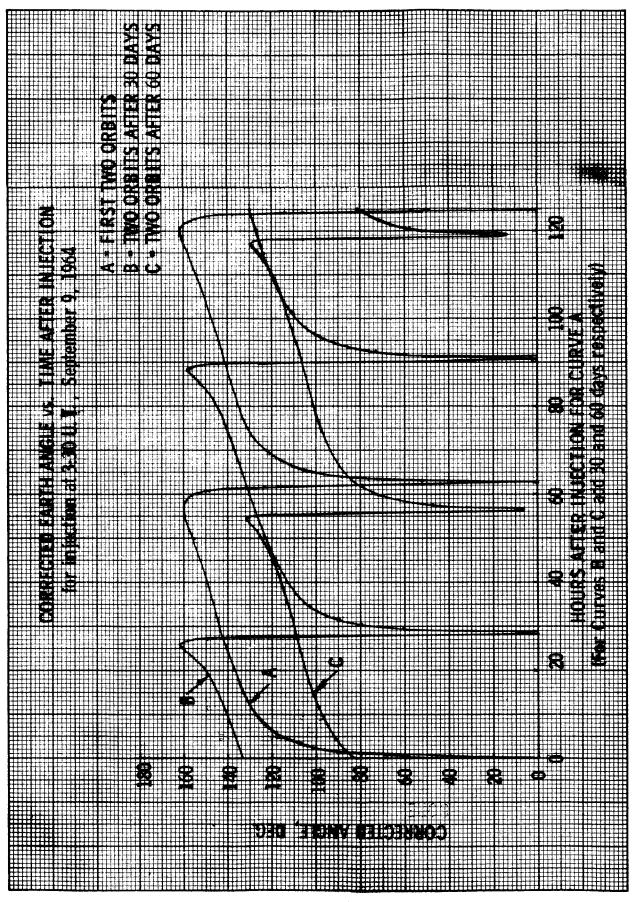
Figure

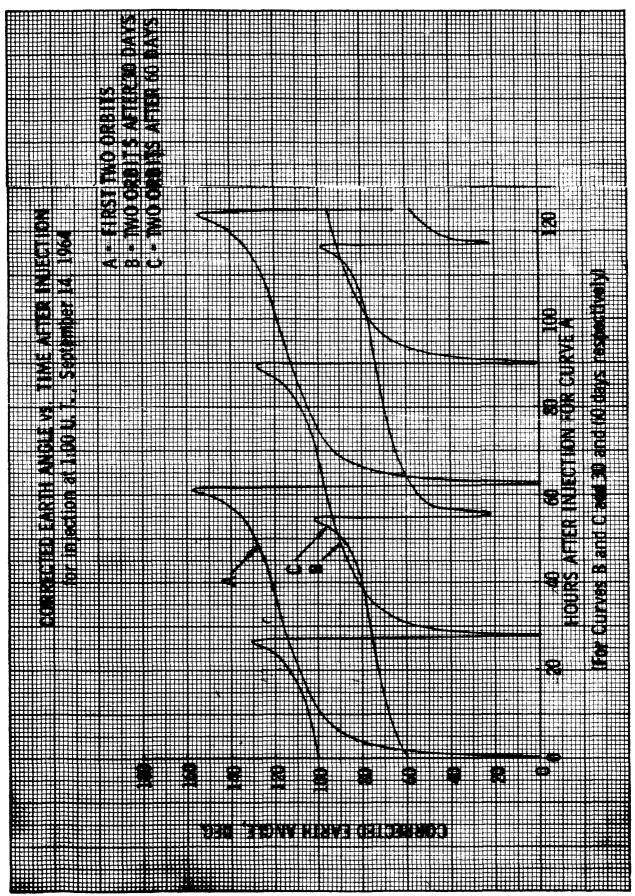
- B-1 Corrected Earth Angle (Injection time, U.T. = 248^d 1^h 0^m 0^s, 1964).
- B-2 Corrected Earth Angle (Injection time, U.T. = 248^d 4^h 0^m 0^s, 1964).
- B-3 Corrected Earth Angle (Injection time, U.T. = 253^d 1^h 0^m 0^s, 1964).
- B-4 Corrected Earth Angle (Injection time, U.T. = $253^{d}3^{h}30^{m}0^{s}$, 1964).
- B-5 Corrected Earth Angle (Injection time, U.T. = 258^d 1^h 0^m 0^s, 1964).
- B-6 Corrected Earth Angle (Injection time, U.T. = 258^d 3^h 0^m 0^s, 1964).
- B-7 Moon Angle (Injection time, U.T. = 248^d l^h 0^m 0^s, 1964).
- B-8 Moon Angle (Injection time, U.T. = 248^d 4^h 0^m 0^s, 1964).
- B-9 Moon Angle (Injection time, U.T. = $253^{d}1^{h}$ 0^m 0^s, 1964).
- B-10 Moon Angle (Injection time, U.T. = 253^{d} 3^{h} 30^{m} 0^{s} , 1964).
- B-11 Moon Angle (Injection time, U.T. = 258^{d} 1^h 0^m 0^s, 1964).
- B-12 Moon Angle (Injection time, U.T. = 258^{d} 3^{h} 0^{m} 0^{s} , 1964).

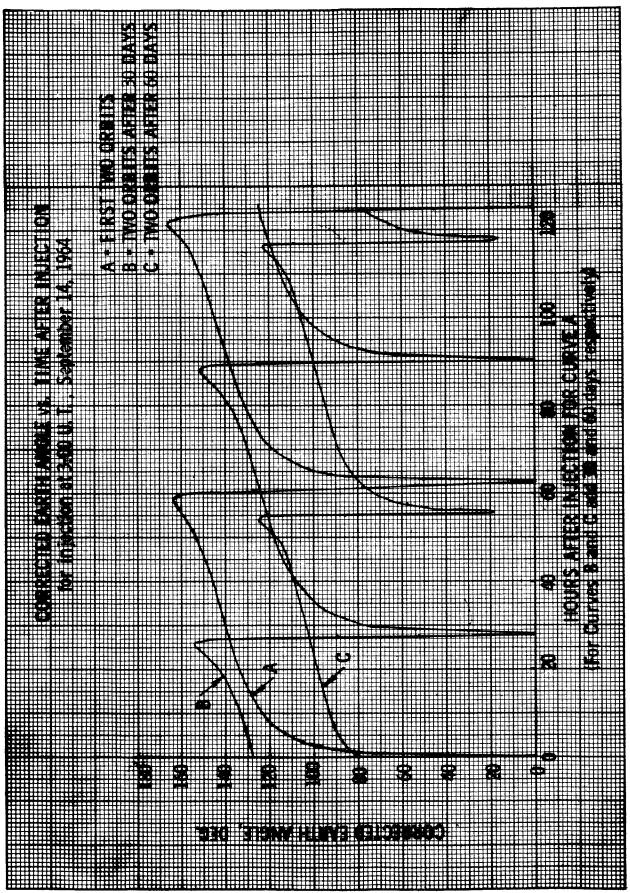


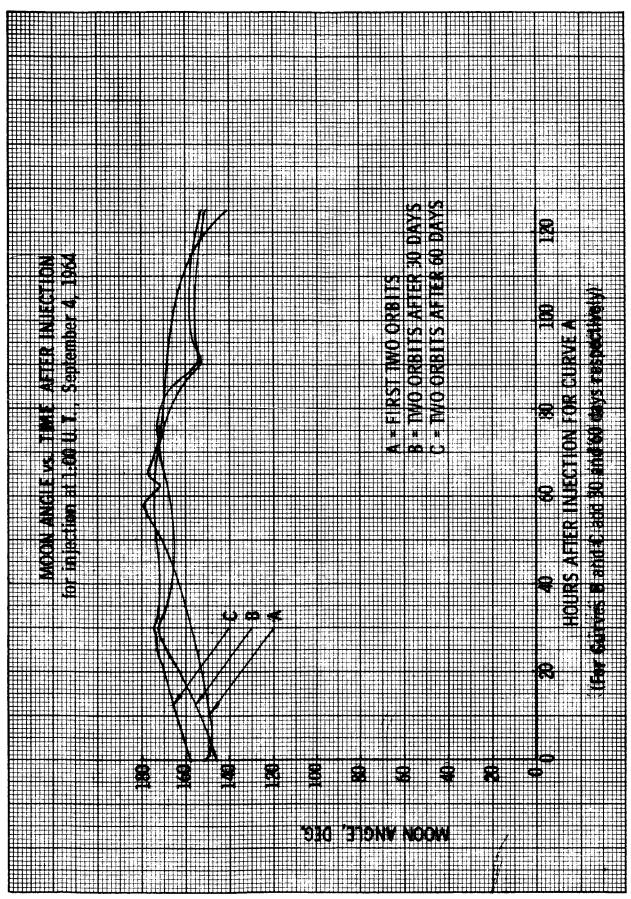


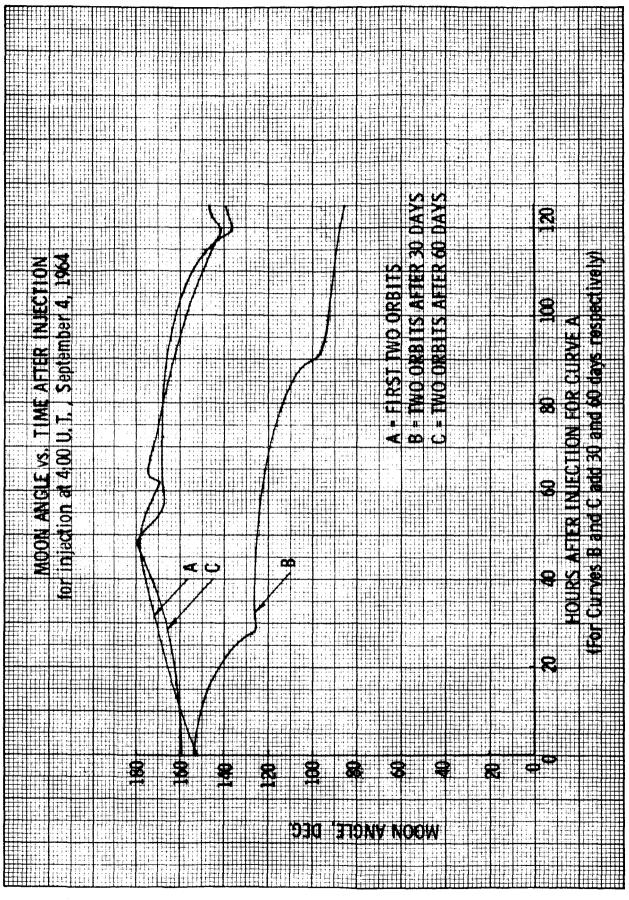


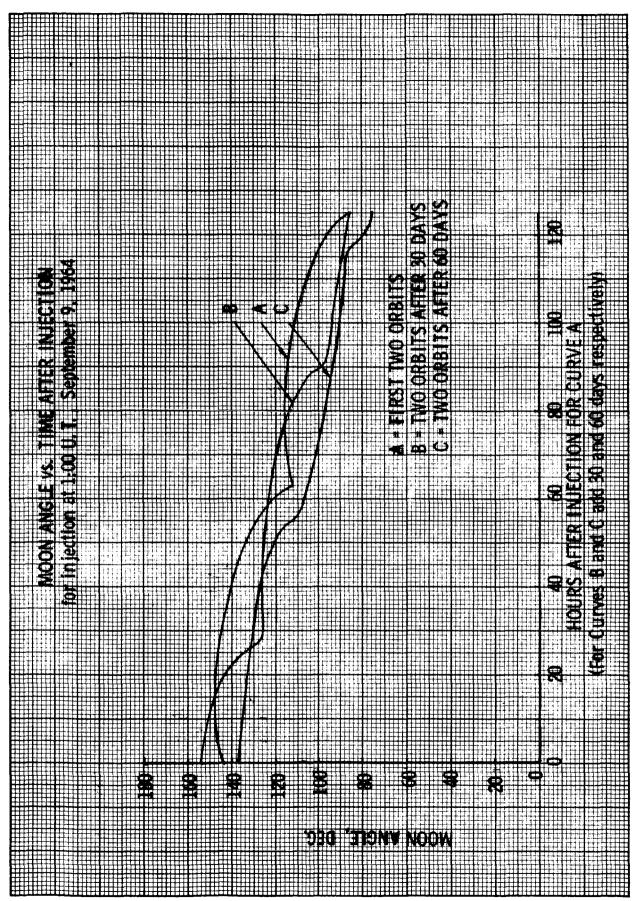


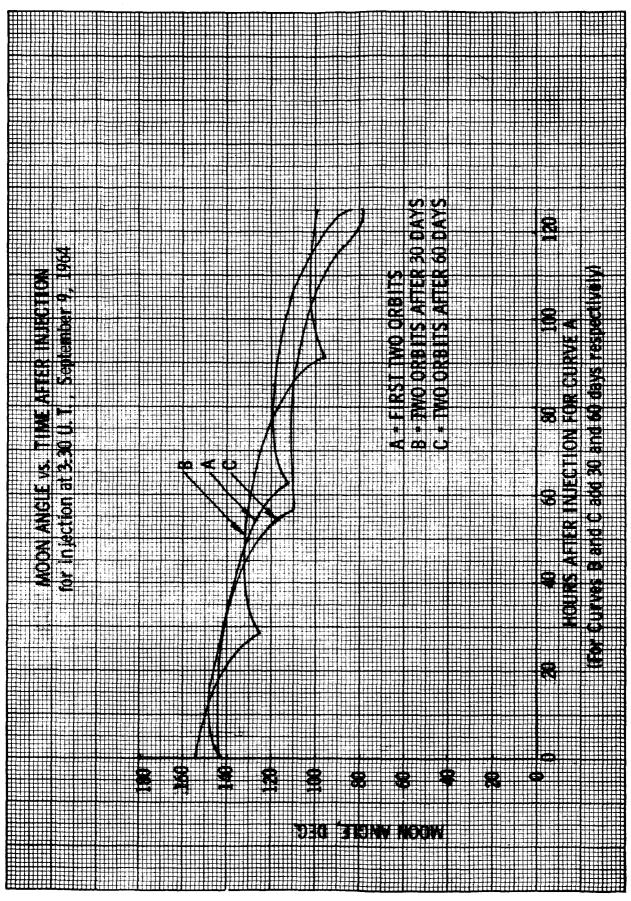


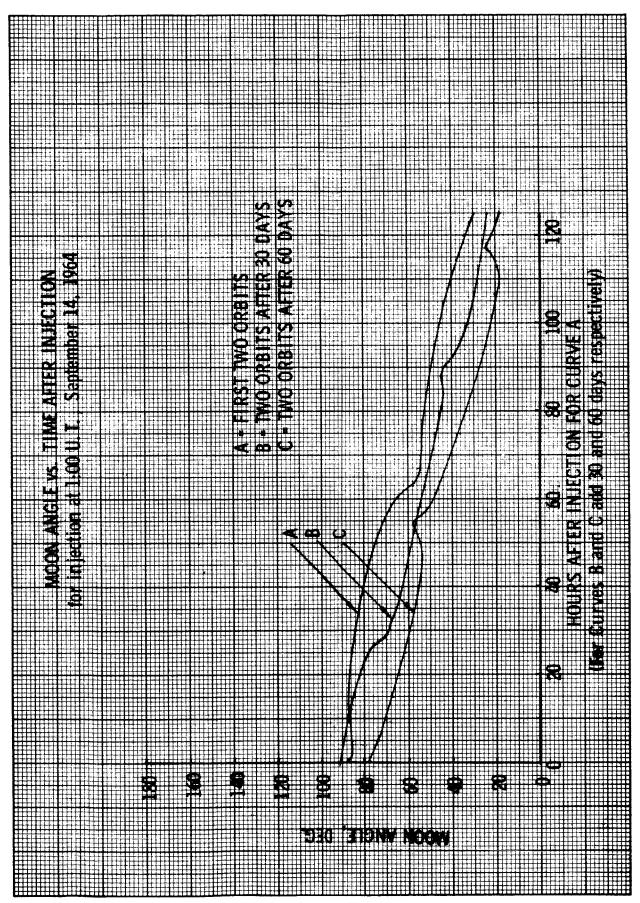


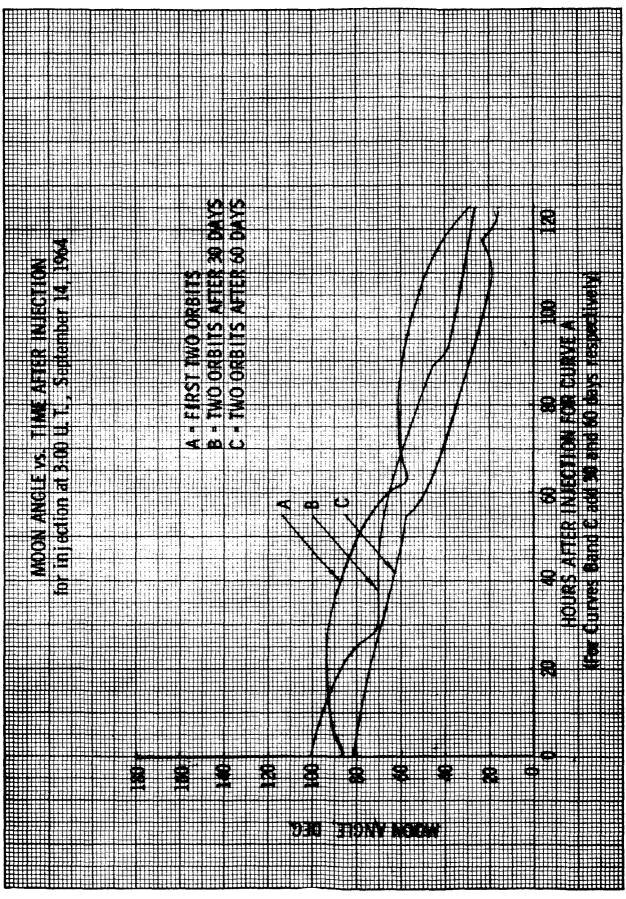












APPENDIX C

SPACECRAFT ANGLES

		76.061	66.171	10.00	100.001	90.40	60.66	710/77
28.000	1 112.43	135.59	127.05	56.19	155.84	37.75	99.14	119.91
29.000	1 112.14	134.85	126.69	56.77	154.88	38.55	98.66	120.11
30.000	1111.85	134.10	126.33	57.36	153.91	39.35	98.17	120.29
31.000	1 111.55	133.35	125.95	57.95	152.94	40.16	19.16	120.46
32,000	1 111.25	132.58	125.56	58.55	151.97	40.98	97.17	120.61
33.000	1 110.93	131.81	125.16	59.16	150.99	41.81	19.96	120.76
34.000	1 110.62	131.03	124.75	59.77	150.01	42.63	96.17	120.89
35.000	1 110.29	130.24	124.32	60.38	149.03	43.47	19.69	121.02
36.000	96.601	129.44	123.89	61.01	148.05	44.31	95.16	121.13
37.000	1 109.62	128.64	123.44	61.63	147.07	45.15	94.65	121.23
38.000	1 109.27	127.84	122.99	62.26	146.09	46.00	94.14	121.31
39.000	1 108.92	127.02	122.52	62.90	145.10	46.86	93.62	121.39
40.000	1 108.57	126.21	122.05	63.54	144.11	47.71	93.11	121.45
41.000	1 108.20	125.38	121.56	64.18	143.12	48.58	92.59	121.50
42.000	1 107.83	124.55	121.07	64.83	142.14	49.44	92.07	121.54
43.000	1 107.46	123.72	120.57	65.48	141.15	50.31	91.55	121.57
44.000	1 107.08	122.88	120.05	66.13	140.15	51.18	91.02	121.59
45.000	1 106.69	122.04	119.54	62.99	139.16	52.05	90.50	121.59
46.000	1 106.30	121.19	119.01	67.45	138.17	52.93	86.68	121.58
47.000	1 105.91	120.34	118.47	68.12	137.18	53.81	89.45	121.56
48.000	1 105.51	119.49	117.93	68.78	136.18	54.69	88.92	121.53
49.000	1 105.10	118.63	117.38	69.45	135.19	55.58	88.40	121.48
50.000	1 104.69	117.77	116.82	70.13	134.19	56.46	87.87	121.42
51.000	1 104.28	116.91	116.26	70.80	133.20	57.35	87.34	121.35
52.000	1 103.86	116.04	115.69	71.48	132.20	58.24	86.81	121.27
53.000	1 103.44	115.17	115.11	72.16	131.21	59.13	86.28	121.18
54.000	1 103.01	114.30	114.53	72.84	130.21	60.03	85.75	121.08
55.000	1 102.58	113.42	113.94	73.53	129.21	60.92	85.22	120.96
600								

DEG.	DEG
15.00000 DEG.	1.00000
= NOISIA	1 ****
ANGLE FIELD OF	I ON SOLAR ARRAY ANGLE ****
LIMIT ON HALF ANGLE FIELD OF VISION	INCREMENT ON SC

ĺ

SOI AR ARRAY								
ANGLE (DEC)	I E.P.1	E.P.2	E.P.3	E.P.4	E.P.5 (EDGE)	E.P.6	0PEP 1	OPEP
1.000	I I 116,96	149.88	131.13	44.02	171.95	22.04	110.70	111-10
2.000	116-911	149.64	131.18	44.30	172.40	22.28	110.34	111.53
3.000	1 1 116.84	149.37	131.22	44.61	172.73	22.55	109.98	6-111
4.000	1 116.77	149.06	131.24	44.93	172.94	22.86	19.601	112.37
5.000	1 116.69	148.74	131.24	45.26	173.01	23.21	109,23	112.78
000.9	I 116.61	148.38	131.23	45.61	172.94	23.59	108.84	धाः ।
7.000	1 116.51	148,00	131.20	45.97	172.72	24.01	108.46	113.58
8.000	1 1 116.40	147.60	131.16	46.35	172.38	24.46	108.06	113.97
000*6	I I 116.28	147.17	131.10	46.74	171,93	24.94	107.66	114,35
10.000	1 116.16	146.72	131.02	41.14	171,38	25.44	107.25	114.73
11.000	1 116.02	146.24	130.93	47.55	170,76	25.97	106.84	115.09
12.000	1 115.88	145.75	130.82	47.98	170.08	26.53	106.43	115.45
13.000	1 115.73	145.23	130.70	48.42	169.34	27.11	100.01	115.80
14.000	1 115.57	144.70	130.56	48.87	168.56	27.71	105.58	116.14
15.000	1 115,40	144,14	130.40	49.33	167,75	28.33	105.15	116.47
16.000	1 115.22	143.57	130.23	49.81	166.92	28.97	104.71	116.79
17.000	1 I 115.03	142.98	130.04	50.29	166.06	29.63	104.27	117.11
18.000	1 I 114.83	142.38	129.84	50.78	165,18	30.31	103.83	117.41
19,000	I I 114.63	141.76	129.63	51.28	164.28	31.00	103.38	117.711
20.000	1 114.42	141.12	129.40	51.80	163.38	31.70	102.92	117.99
21.000	I I 114.20	140•47	129,15	52,32	162.46	32.42	102.46	118.27
22.000	I I 113•97	139.81	128.89	52.85	161.54	33.15	102.00	118.54
23.000	1 1 113.73	139.14	128.62	53.38	160.60	33,89	101.53	118,79
24.000	1 113.48	138.45	128.33	53.93	159.66	34.64	101.06	119,04
25,000	1 113.23	137.75	128.03	54.48	158,71	35.41	100.59	119.27
	ı							

57.000 I	58.000	59.000	60.000	61.000	62.000 1	63.000	64.000	65.000	000.99	67.000	68.0C0 I	69.000	70.000	71.000	72.000 1	73.000 1	74.000 1	75.000 1	76.000	77.000	78.000	79.000 1	80.000 1	81.000	82.000 1	83.000 I	84.000	85.000 I	7
101.71	101.27	100.82	100.38	99.93	74.66	99.02	98.56	98.10	97.63	97.17	96.70	96.23	95.76	95.29	94.81	94.34	93.86	93.38	92.90	92.42	91.94	91.46	90.98	90.50	90.02	89.54	89.06	88.58	
1111.66	110.78	109.90	109.01	108.12	107.23	106.34	105.44	104.55	103.65	102.76	101.86	100.96	100.06	99.16	98.25	97.35	96.45	95.54	94.64	93.74	92.83	91.93	91.02	90.11	89.21	88.30	87.40	86.49	
112.74	112.13	111.52	110.91	110.29	109.66	109.03	108.40	107.76	107.12	106.47	105.83	105.17	104.52	103.86	103.20	102.54	101.88	101.21	100.54	78.66	99.20	98.52	97.84	97.17	64.96	95.81	95.13	94.45	
14.90	75.59	76.29	76.98	77.67	78.37	79.07	17.67	80.47	81.17	81.87	82.57	83.28	83.98	84.69	85.39	86.10	86.80	87.51	88.22	88.93	89.63	90.34	91.05	91.75	92.46	93.17	93.88	94.58	
127.22	126.22	125.22	124.22	123.22	122.22	121.22	120.22	119.22	118.22	117.22	116.22	115.22	114.22	113.22	112.22	111.21	110.21	109.21	108.21	107.21	106.21	105.21	104.20	103.20	102.20	101.20	100.20	99.20	
62.72	63.62	64.52	65.43	66.33	67.24	68.14	69.05	96.69	70.87	71.78	72.69	73.60	74.52	75.43	76.34	77.26	78.17	79.09	80.01	80.92	81.84	82.76	83.68	84.59	85.51	86.43	87.35	88.27	
84.16	83.64	83.11	82.58	82.06	81.53	81.01	80.49	19.97	19.45	78.93	78.42	77.91	77.40	76.89	76.39	75.89	75.39	74.90	74.41	73.92	73.43	72.96	72.48	72.01	71.54	71.08	70.63	70.17	
120.70	120.55	120.39	120.22	120.04	119.84	119.64	119.43	119.20	118.97	118.73	118.47	118.21	117.94	117.66	117.37	117.07	116.77	116.45	116.13	115.80	115.46	115.12	114.77	114.41	114.04	113.67	113.29	112.90	

92.40 96.70 96.19 91.03 68.85- 111.71	91,72 97,40 95,19 91,95 68,43 111,30	91.04 98.10 94.19 92.87 68.00 110.89	90.36 98.81 93.19 93.79 67.59 110.47	89.68 99.51 92.18 94.70 67.18 110.05	88.99 100.21 91.18 95.62 66.78 109.62	100.91 90.18 96.54 66.38 109.19	87.63 101.61 89.18 97.46 65.99 108.75	86.95 102.30 88.18 98.38 65.61 108.31	86.27 103.00 87.18 99.30 65.24 107.86	85.60 103.69 86.18 100.22 64.87 107.41	84.92 104.38 85.18 101.13 64.52 106.96	84.25 105.07 84.18 102.05 64.17 106.50	83,57 105,76 83,18 102,97 63,83 106,04	82.90 106.45 82.18 103.89 63.49 105.58	82-23 107-13 81-18 104-80 63-17 105-11	81.55 107.82 80.18 105.72 62.86 104.65	80.90 108.50 79.18 106.63 62.55 104.17	80.23 109.18 78.18 107.55 62.26 103.70	79.57 109.85 77.18 108.46 61.97 103.22	110.53 76.18 109.37 61.69 102.75	78.26 111.20 75.18 110.29 61.43 102.27	77,60 111,86 74,18 111,20 61,17 101,78	76.95 112.53 73.18 112.11 60.92 101.30	76,30 113,19 72,18 113,02 60,69 100,81	75.66 113.85 71.19 113.93 60.46 100.33	75.02 114.50 70.19 114.84 60.25 99.84	74.38 115.16 69.19 115.74 60.05 99.35	
83.78	82.88	81,98	81.08	80.18	79,28	78,38	77.48	76.59	75.69	74.80	73.91	73,02	72.13	71.25	70.36	84.69	68.60	67,72	66.85	65.97	65.10	64.24	63,37	62,51	61.65	60.80	56.65	
1 87414	I I 86.66	I I 86.19	1 85.71	I 85.24	1 84-77	I 84.29	I I 83.83	1 1 83,36	I 82.89	I 82,43	I I 81.97	I I 81.51	1 81.06	1 80.60	1 80.15	1 79.71	1 79.26	1 78.82	I 78.38	1 17,95	1 77.52	1 77.09	1 76.67	1 76.25	I 75.84	I 75,43	I I 75.02	
88,000	000*68	000*06	91.000	92.000	93.000	94.000	95.000	000*96	97.000	98.000	000*66	100.000	101.000	102,000	103.000	104.000	105.000	106.000	107.000	108.000	109.000	110.000	111.000	112,000	113.000	114.000	115.000	

88.04 87.55 87.55 86.59 86.11 85.16 85.16 84.69 84.69
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65.04 35.67 64.86 35.13 64.86 35.13 64.69 34.61 64.53 34.12 64.24 33.18 64.24 33.18 64.24 33.18 64.24 33.18 64.24 33.18 63.48 31.25 63.48 31.25 63.48 30.65 63.48 30.40 63.41 30.40 63.41 30.40 63.41 30.40 63.41 30.40 63.41 30.40 63.41 30.40 63.41 30.40 63.18 29.56 63.16 29.43 63.15 29.43 63.15 29.43 63.15 29.43 63.15 29.48 63.15 29.48 63.15 29.48 63.27 29.48 63.27 29.48 63.27 29.80 63.27 29.80 63.27 29.80 63.27 29.80 63.27 29.80 63.27 29.80 63.27 29.80 63.27 29.80 63.27 29.80 63.23 29.96 63.23 29.96 63.23 29.96 63.23 29.96 63.23 29.96 63.23 29.96 63.23 29.96 63.23 29.96 63.23 29.80 63.23 29.96 63.23 29.80		99			32																138.10 17.38		138.19 15.58	*	*	•	*	•	137.99 * 10.62
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69.59 47.70 55.29 125.74 25.76 142.30 84.33 62.47 69.88 48.43 55.65 125.14 26.71 141.69 84.83 62.31 70.18 49.17 56.03 124.54 27.67 140.67 85.33 62.15 170.49 49.91 56.42 123.31 29.58 139.00 86.33 61.87 170.49 49.91 56.42 123.31 29.58 139.00 86.33 61.87 171.11 51.41 57.21 122.68 30.54 138.16 86.83 61.73 171.15 52.17 57.62 122.04 31.51 137.31 87.33 61.49 171.76 52.93 58.04 121.40 32.47 136.45 87.83 61.49 171.76 52.93 58.04 121.40 32.47 136.45 88.33 61.38 120.09 34.40 134.73 88.82 61.28 120.45 55.24 59.36 120.09 34.40 134.73 88.82 61.19 123.10 56.02 59.80 118.75 36.34 132.10 90.31 61.03 173.79 57.58 60.72 118.38 38.28 131.22 90.80 60.96 6	60.72 117.39 38.28 131.22 90.80 61.19 116.70 39.25 130.34 91.30 61.66 116.01 40.22 128.55 92.28 62.63 114.60 42.16 127.66 92.76 63.13 113.89 43.14 126.76 93.25 64.65 111.74 46.06 124.95 94.22 64.65 111.74 46.06 124.95 94.22 65.69 110.28 48.99 121.31 96.12 65.69 110.28 48.99 121.31 96.12 66.74 108.80 49.97 120.39 96.59	107.31 51.92 118.55 97.53 60.90 106.56 52.90 117.63 97.99 60.96
47.70 55.29 125.74 25.76 142.30 48.43 55.65 125.14 26.71 141.49 49.17 56.03 124.54 27.67 140.67 49.91 56.42 123.93 28.63 139.84 50.66 56.81 123.31 29.58 139.00 51.41 57.21 122.04 30.54 138.16 52.17 57.62 122.04 31.51 137.31 52.93 58.04 121.40 32.47 136.45 53.69 58.47 120.75 33.44 135.59 54.46 58.90 120.09 34.40 134.73 55.24 59.35 119.43 35.37 133.86 56.02 59.80 118.75 36.34 132.98 56.80 60.25 118.08 37.31 132.98 56.80 60.25 118.08 37.31 132.98 57.58 60.72 117.39 38.28 131.22	61.19 116.70 39.25 130.34 61.16 116.70 39.25 130.34 61.66 116.01 40.22 129.45 62.63 114.60 42.16 127.66 63.13 113.89 43.14 125.85 64.14 112.46 45.09 124.95 64.65 111.01 47.04 123.13 65.69 110.28 48.01 122.22 66.21 109.54 48.99 121.31 66.74 108.80 49.97 120.39 65.78 108.06 50.94 119.47	51.92 118.55
47.70 55.29 125.74 25.76 48.43 55.65 125.14 26.71 49.17 56.03 124.54 27.67 49.91 56.42 123.93 28.63 50.66 56.81 123.93 28.63 51.41 57.21 122.68 30.54 52.17 57.62 122.04 31.51 52.17 57.62 122.04 31.51 52.93 58.04 121.40 32.47 54.46 58.90 120.09 34.40 55.24 59.35 119.43 35.34 56.02 59.80 118.75 36.34 56.80 60.25 118.08 37.31 56.80 60.25 117.39 38.28 57.58 60.72 117.39 38.28	60.72 117.39 38.28 61.19 116.70 39.25 61.66 116.01 40.22 62.14 115.31 41.19 62.63 114.60 42.16 63.13 113.89 43.14 63.63 113.18 44.11 64.16 112.46 45.09 64.65 111.01 47.04 65.69 110.28 48.01 66.21 109.54 48.99 66.74 108.80 49.97	51.92
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41.10 55.29 48.43 55.65 49.17 56.03 49.91 56.42 50.66 56.81 51.41 57.21 52.17 57.21 52.93 58.47 53.69 58.47 54.46 58.90 55.24 59.80 56.02 59.80 56.80 60.25 57.58 60.25	60.72 61.19 61.66 62.14 62.63 63.13 64.16 65.16 65.16 65.21 66.21	7.31
47.70 55.29 48.43 55.65 49.17 56.03 49.91 56.42 50.66 56.81 51.41 57.21 52.17 57.21 52.17 57.62 52.17 57.62 52.17 57.62 52.93 58.47 54.46 58.90 55.24 59.35 56.80 60.25		91
	35 35 35 35 35 35 35 35 35 35 36 37 37 37 37	68.37
69.59 1 69.88 1 70.18 1 70.49 1 71.11 1 71.43 1 71.43 1 72.09 1 72.09 1 73.10 1 73.10	51 58 59 59 59 60 60 60 60 60 60 60 60 60 60 60 60 60	68.83
·┩╾┥╾┥╾┩╾┥╾┥╾┩╾┩╼╡ ╸ ┩╾┥ ╸ ┥╾┥	74.14 74.50 74.50 74.86 75.22 75.22 75.29 75.96 76.33 76.10 77.85 77.85 78.62	79.40
209,000 210,000 211,000 212,000 214,000 215,000 215,000 216,000 216,000 217,000 217,000 217,000 227,000	222,000 1 224,000 1 225,000 1 225,000 1 229,000 1 229,000 1 239,000 1 239,000 1 233,000 1 235,00	236.000 I I 237.000 I

61.10	61-19	61.28	61.38	61.50	- 61-62	61.75	61.88	62.03	62.18	62.35	62.52	62.70	62.88	63.08	63-28	63.50	. 63.12.	63.95	64.18	64.43	64.68	64.94	65.21	65.48	65.77	90-99	66.35	99-99	76.93
98,91	99,37	99.82	100.27	100.72	101.16	101.60	102.03	102.47	102.90	103.32	103.74	104.16	104.58	104.99	105,39	105.79	106.19	106.58	106.97	107.36	107.74	108.11	108.48	108.85	109.21	109.56	10.601	110.26	110.60
115.78	114.86	113.93	113.00	112.07	111.14	110.21	109.27	108.34	107.40	106.47	105.53	104.59	103,65	102.72	101.78	100.84	99.90	98-96	98.01	70.79	96.13	95.19	94.25	93.31	92.36	91.42	90.48	89.54	88.59
54.86	55.83	56.81	57.79	58.77	59.75	60.73	61.71	62.69	53.67	64.65	65.63	19.99	67.59	68.57	69.55	70.54	11.52	72.50	73.48	74.46	75.45	76.43	17.41	78.39	79.37	80.36	81.34	82.32	83.31
105.06	104.30	103.54	102 - 78	102.01	101.25	100.48	99.71	98.94	98.16	97.39	19•96	95.84	95.06	94.28	93.50	92.72	91,94	91-16	90.38	89.60	88.82	88.04	87.26	86.48	85.70	84.92	84.15	83.37	82.59
74.69	70.03	70.59	71.15	71.72	72.29	72.86	73.44	74.02	14.61	75.19	75.78	76.38	76.97	77.57	78.17	78.77	79.37	79.98	80.58	81.19	81.80	82.41	83.03	83.64	84.26	84.88	85.50	86.12	86.74
71.29	72,11	72,94	73.76	74.59	75.41	76.24	77.07	17.90	78.73	19.56	80.39	81,22	82.06	82.89	83.72	84.56	85.39	86.23	87.06	87.90	88.73	89.57	90.40	91.24	92.07	92.91	93.75	94.58	95.42
80.20	09*08	81.00	81.40	81.81	82.21	82.62	83.03	83.45	83.86	84.27	69*58	85.10	85.52	85.94	86.36	86.78	87.20	87.62	88.05	88.47	68*88	89.31	89.74	90.16	65.06	91.01	91.43	91.86	92.28
239,000	240.000	241.000	242,000	243.000	244.000	245.000	246.000	247,000	248,000	249.000	250.000	251.000	252,000	253,000	254.000	255.000	256.000	257.000	258,000	259.000	260.000	261.000	262.000	263.000	264.000	265.000	266.000	267.000	268.000 I
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	I 92.70	96.25	87.36	81.82	84.29	87.65	110.93	67.29
270.000	1 93.13	97.09	87.98	81.05	85.27	86.71	1111.26	67.62
271.000	1 93.55	97.92	88.60	80.28	86.26	85.77	111.59	67.95
272.000	1 93.97	98.76	89.23	19.61	87.24	84.83	111.90	68.29
273.000	1 94.39	65.66	89.85	78.74	88.22	83.89	112.22	68.63
274.000	1 94.81	100.42	90.48	17.97	89.21	82.95	112.52	66*89
275.000	1 95.23	101.26	91.10	17.21	90.19	82.01	112.82	69.35
276.000	1 95.65	102.09	91.73	76.45	91.17	81.07	113.12	69.71
277.000	1 96.07	102.92	92.35	15.69	92.16	80.13	113.41	70.09
278.000	1 96.48	103.75	95.98	74.93	93.14	79.19	113.69	70.46
279.000	06.96	104.58	93.60	74.18	94.13	78.25	113.97	70.85
280.000	1 97.31	105.41	94.23	73.42	95.11	77.31	114.24	71.24
281.000	1 97.73	106.24	94.85	72.68	96.10	76.38	114.50	71.64
282.000	1 98.14	107.01	95.48	71.93	97.08	15.44	114.76	72.04
283.000	1 98.55	107.90	96.10	71.19	90.86	74.50	115.01	72.45
284.000	96.86	108.72	96.73	70.45	90.66	73.57	115.26	72.86
285.000	96.96	109.55	97.35	69.71	100.03	72.64	115.50	73.28
286.000	1 99.77	110.37	16.16	86.89	101.02	71.70	115.73	73.70
287.000	1 100.17	1111.19	98.59	68.25	102.00	77.07	115.95	74.13
288.000	1 100.57	112.02	99.21	67.53	102.99	69.84	116.17	74.56
289,000	1 100.97	112.83	69*83	66.81	103.97	68.91	116.38	75.00
290.000	1 101.37	113.65	100.45	66.10	104.96	67.98	116.58	75.45
291.000	1 101.76	114.47	101.07	62.39	105.94	67.05	116.78	75.90
292.000	1 102.15	115.28	101.69	64.68	106.93	66.13	116.97	76.35
293.000	1 102.54	116.10	102.30	63.98	107.91	65.20	117.15	76.81
294-000	1 102.93	116.91	102.91	63.29	108.90	64.28	117.33	17.27
295.000	1 103.31	117.72	103.52	62.60	109.89	63.36	117.49	17.73
296.000	1 103.69	118.52	104.13	61.92	110.87	62.44	117.65	78.20
297.000	1 104.07	119,33	104.74	61.24	111.86	61.52	117.81	78.68
298.000	1 104.45	120 12	00.	1				

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19.64	80.12	80.61	81.10	81.60	82.10	82.60	83.10	83.61	84.12	84.63	85.15	85.66	86.18	86.70	87.22	87.75	88.27	88.80	89.33	89.86	90.39	90.92	91.45	91.98	92.52	93.05	93.58	94.12	94.65	95.18
118.09	118.22	118.34	118.46	118.56	118.66	118.75	118.83	118.91	118.97	119.03	119.08	119.12	119.15	119.18	119.19	119.20	119.20	119.19	119.17	119.15	119.11	119.07	119.02	118.96	118.89	118.82	118.73	118.64	118.53	118.42
59.69	58.78	57.87	96.95	90.95	55.15	54.25	53.35	52.46	51.57	50.68	64.79	48.91	48.03	47.16	46.29	45.43	44.57	43.71	42.86	42.02	41.18	40.35	39.52	38.71	37.90	37.10	36.30	35.52	34.75	33.98
113.83	114.81	115.80	116.78	117.77	118.75	119.74	120.73	121.71	122.70	123.68	124.67	125.65	126.64	127.62	128.61	129.59	130.58	131.56	132.54	133.53	134.51	135.49	136.48	137.46	138.44	139.42	140.40	141.38	142.36	143.34
29.90	59.25	58.60	57.95	57.32	56.69	56.07	55.46	54.86	54.26	53.68	53.11	52.54	51.99	51.44	50.91	50.39	49.88	49.38	48.90	48.43	47.97	47.52	47.09	46.68	46.28	45.89	45.52	45.16	44.82	44.50
105.95	106.55	107.15	107.75	108.34	108.93	109.52	110.10	110.68	111.26	111.83	112.40	112.97	113.53	114.08	114.64	115.18	115.73	116.27	116.80	117.32	117.85	118.36	118.87	119.37	119.87	120.36	120.84	121.31	121.78	122.24
120.93	121.72	122.52	123.31	124.09	124.88	125.66	126.43	127.21	127.97	128.74	129.50	130.25	131.00	131.74	132.48	133.21	133.93	134.65	135.36	136.06	136.75	137.44	138.11	138.78	139.44	140.08	140.72	141.34	141.95	142.54
1 104.82	105.19	105.56	105.92	106.28	106.63	106.99	107.33	107.68	108.02	108.35	108.68	109.01	109.33	109.65	109.96	110.27	110.57	110.87	1111.16	111.45	111.73	112.00	112.27	112,53	112.79	113.04	113.28	113.52	113.75	113.98
299.000	300.000	301.000	302.000	303.000	304.000	305.000	306.000	307.000	308.000	309.000	310.000	311.000	312.000	313.000	314.000	315.000	316.000	317.000	318.000	319.000	320.000	321.000	322.000	323.000	324.000	325.000	326.000	327.000	328.000	329.000 1

95.71	96.25	96.78	97.31	97.84	98.36	98.89	99.42	76*66	100.46	100.98	101.50	102.01	102.52	103.03	103.54	104.04	104.54	105.04	105.53	106.02	106.51	106.99	107.47	107.94	108.41	108.87	109.33	109.78	110.23	110.67
118.30	118,18	118.04 94	9 06.711	117.74	117.58 9	117.42 9	117.24	117.05	116.86 10	116.66 10	116.45 10	116.23 10	116.01 10	115.77 10	115.53 10	115.29 10	115.03 10	114,77 10	114.50 10	114.22 10	113.93 10	113.64 10	113.34 10	113.04 10	112.72 10	112.40 10	112.08 10	111.74 10	111.40	111.05
33.23	32.49	31.17	31,05	30.36	29.67	29.01	28-36	27.13	27.13	26.54	25,98	25.45	24.94	24.46	24.01	23.59	23.21	22-86	22.54	22,27	22,03	21.84	21.69	21.58	21.51	21.49	21.52	21.58	21.69	21.85
144,32	145.30	146.27	147.25	148.22	149.20	150-17	151-14	152,10	153.07	154.03	154.99	155.95	156.91	157.86	158.80	159.74	160.68	161.61	162.53	163.44	164.34	165.23	166.11	166,96	167.79	168.60	169.37	170-11	170.79	171.41
44.19	43.91	43.63	43.38	43.15	42.93	42.73	42.55	42.39	42,25	42.13	42.02	41.94	41.88	41.84	41.81	41.81	41.82	41.86	41.91	41.99	42.08	42.19	42.32	42.47	45.64	42.83	43.03	43.25	43.49	43.75
122.69	123,13	123.56	123.99	124.40	124.81	125,20	125.59	125,96	126,33	126.68	127.03	127.36	127.68	127.99	128.28	128.56	128.83	129-09	129.33	129.56	129.78	129,98	130-17	130,34	130.50	130.64	130.77	130.89	130.98	131.07
143.12	143.69	144.24	144.78	145.30	145.80	146.28	146.74	147.18	147.60	148.00	148.37	148-72	149,05	149.35	149.62	149.86	150.08	150-27	150.42	150.55	150.65	150.71	150.74	150,75	150.72	150.65	150.56	150.44	150.28	150.10
114.19	114.40	114.61	114.80	114.99	115.17	115,35	115.51	115.67	115,82	115.96	116.10	116.22	116.34	116.45	116.55	116-64	116.73	116.80	116.87	116.93	116.97	117.01	117.04	117.07	117.08	117.08	117.07	117,06	117,03	117,00
1 _ 000°088	I 000 188	1 332.000 1	1 333,000 1	1 000 7 7 E	1 335,000 I	I 336.000 I	337,000 1	338,000 I	1 000.955	1 340.000 I	341.000 1	1 342,000 I	1 343,000 I	1 344.000 I	1 345.000 I	1 346.000 1	347.000 1	1 348.000 I	1 349.000 I	350.0C0 I	1 351,000 1	352.000 I	353.000 I	354.000 I	355.000	356,000 I	357.000 I	1 358,000 I	1 359.000 1	360.000 I
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